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Response variance prediction using transient statistical energy analysis

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Abstract

26 Statistical Energy Analysis (SEA) is a prominent method for predicting the high frequency
27 response of complex structures under steady loading where the structure is split into subsystems
28 and the subsystem energies are calculated. Since at high frequencies, the dynamic response of
29 nominally identical structures can differ greatly, methods have been developed to predict both
30 the mean and variance of the energy in the subsystems of a system across an ensemble of
31 systems. SEA can be extended to predict the transient response of a system, either to shock or
32 time-varying inputs and is known as Transient SEA, although this formulation has so far only
33 been interested in the mean response. In this paper, a method for predicting the variance of the
34 transient response is derived by considering how an individual realisation can deviate from the
35 mean. A matrix differential equation for the covariance of the subsystem energies is derived
36 which is driven by terms representing the variability in the system. These variance terms are
37 provided by assuming that the natural frequencies in each subsystem conform to the Gaussian
38 Orthogonal Ensemble. The accuracy of the method is investigated both numerically and
39 experimentally using systems involving coupled plates and its limitations are discussed.

40

41 *Keywords:* TSEA; Transient vibration; Shock-induced vibration; SEA Variance

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45 **I. Introduction**

46 The transient response of structures to either shock or time-varying loading is of considerable
47 importance when designing systems to protect against failure or reduce noise. Transient
48 Statistical Energy Analysis (TSEA) (see for example Lai and Soom, 1990b, 1990a; Langley et
49 al., 2019; Lyon and DeJong, 1995; Manning and Lee, 1968; Pinnington and Lednik, 1996a,
50 1996b) is a popular technique used to predict the high frequency response of complex structures
51 under these conditions and has been applied to both academic and industrial applications such
52 as predicting the noise levels in buildings due to impulses and footsteps (Robinson and
53 Hopkins, 2015).

54 Transient SEA is based on steady-state SEA which has been developed over a number of
55 decades to predict the response of complex structures subjected to steady external loads.
56 Steady-state SEA is most commonly employed at high frequencies where, due to the short
57 wavelength of vibrations, traditional vibration analysis approaches such as the finite element
58 method become undesirable. At these frequencies and wavelength-scales, an extremely
59 detailed model with a very fine mesh is required and any imperfections in the system of
60 comparable length-scale to the wavelength will have a significant effect on the structural
61 response. Consequently, nominally identical structures can generate very different frequency
62 response characteristics. The SEA method inherently circumvents these issues by splitting a
63 structure into large regions that contain similar properties known as subsystems and concerning
64 itself only with the average and variance of the energy in each subsystem where the average is
65 taken both spatially over the subsystem and over an ensemble of nominally identical
66 subsystems that encompasses all random imperfections.

67 In addition to the mean energy of each subsystem, it is important to quantify the spread of the
68 energy within each system across the ensemble since any particular system drawn from the

69 ensemble may have subsystem energies that are very different from the ensemble average
70 values. A preliminary method for calculating the response variance was suggested by Lyon and
71 DeJong (1995), but a more suitable and widely adopted method was later developed by Langley
72 and Cotoni (2004a). This built on the work by Langley and Brown (2004a, 2004b) that used
73 the principle of universality, which states that if a system is sufficiently random then the
74 statistical distribution of its natural frequencies is independent of how the system is
75 randomised. Assuming, due to strong empirical evidence, see for example (Weaver, 1989), that
76 the natural frequencies of a subsystem are governed by the Gaussian Orthogonal Ensemble
77 (GOE) Mehta (2004) and that the modeshapes are Gaussian across the ensemble, the variance
78 of the energy in a built-up system can be calculated by investigating how any single realisation
79 differs from the mean (Langley and Cotoni, 2004a).

80 It is thought that the extension of steady-state SEA to the transient case is reasonable when the
81 assumptions on which steady-state SEA are built are not violated. This generally requires the
82 energy in the system to vary slowly with time compared to the time period of its oscillations
83 and many modes to be excited. Under these conditions, the steady-state SEA coupling loss
84 factors (CLFs) can be used for the transient analysis and have been shown to produce strong
85 results (Hopkins and Robinson, 2013; Langley et al., 2019; Robinson and Hopkins, 2015)
86 despite suggestions that time-varying CLFs could be more suitable, particularly at lower
87 frequencies (Lai and Soom, 1990b).

88 At present, there exist well-tested SEA-based methods for calculating the mean and variance
89 of the response of complex systems to steady loading and the mean response to transient and
90 impulsive loading. This paper aims to derive a method for calculating the variance of the
91 response under shock and transient external loading conditions using a similar approach to
92 Langley and Cotoni (2004a). In what follows Section II presents a brief summary of the TSEA
93 equations used to calculate the mean response before applying them in Section III to derive

94 equations for calculating the variance of the response. Numerical and experimental validation
 95 using systems of coupled plates is then presented in Sections IV and V respectively before the
 96 limitations of the method are discussed in Section VI and conclusions are made in Section VII.

97

98 **II. Summary of the TSEA mean equations**

99 In this section, a brief summary of the TSEA method for calculating the mean subsystem
 100 energies is presented. The TSEA equation is based on the steady-state SEA equation derived
 101 by Lyon and DeJong (1995) by considering a power balance between subsystems at each
 102 frequency such that

$$103 \quad \mathbf{P}_{\text{in}}(\omega) = \mathbf{A}(\omega)\mathbf{E}(\omega), \quad (1)$$

104 which can be written using index notation as

$$105 \quad P_{\text{in},j} = \omega \eta_j E_j + \sum_{k=1}^N \omega \eta_{jk} n_j \left(\frac{E_j}{n_j} - \frac{E_k}{n_k} \right) \quad (2)$$

106 where ω is the frequency of interest, E_j , n_j , η_j and $P_{\text{in},j}$ are the ensemble average vibrational
 107 energy, modal density, internal loss factor and power into subsystem j and η_{jk} is the coupling
 108 loss factor between subsystems j and k . If the system is subjected to a shock or time-varying
 109 external power input, a time derivative term can be included in the SEA equation representing
 110 an increase or decrease of energy in the subsystems with time. However, this means the
 111 subsystem energies are a function of both time and frequency. Most commonly, the TSEA
 112 equation is averaged over a frequency band, $\omega_1 < \omega < \omega_2$ say, such that the energy variable
 113 represents the time evolution of the ensemble average of the energy in a given band. The TSEA
 114 equation is thus

115
$$\mathbf{P}_{\text{in}}(t) = \frac{d\mathbf{E}(t)}{dt} + \mathbf{A}\mathbf{E}(t) \quad (3)$$

116 where each term is now considered to be averaged over the frequency band. This equation is
117 used to derive the equations for predicting the variance of the subsystem energies.

118 Much previous work has addressed the conditions under which the steady state SEA equations,
119 Eq. (1), are valid, and these conditions can be expressed in terms of either modes or waves. In
120 modal terms, the modal responses of each subsystem must be uncorrelated and have
121 equipartition of energy, and they must also be uncorrelated from the modal responses of other
122 subsystems. An equivalent statement in terms of waves is that each subsystem must carry a
123 diffuse wavefield (in an ensemble sense), and the wavefields must be uncorrelated across the
124 different subsystems. These conditions are promoted by weak coupling between the
125 subsystems. Recent work on the mean TSEA, Eq. (3), has elucidated the additional conditions
126 which must be met for these equations to be valid (Langley et al., 2019). In wave terms, the
127 rate of change of energy within a subsystem must be slow compared to the time taken for a
128 wave to transit the subsystem several times, so that the subsystems can be considered to carry
129 diffuse wavefields at all times. This condition is again promoted by weak coupling between
130 the subsystems. The following section considers the variance of the response, and this requires
131 the additional assumption that the natural frequencies and mode shapes of each subsystem have
132 statistics which are governed by the Gaussian Orthogonal Ensemble. As discussed by Wright
133 and Weaver, 2010, this condition is widely applicable to structures which display a sufficient
134 degree of randomness, in the sense that random variations in the natural frequencies are greater
135 than the mean modal spacing (Kessissoglou and Lucas, 2009).

136

137

138 **III. Derivation of the TSEA variance equations**

139 This section derives the TSEA equations for predicting the time-varying variance of the system
 140 response based on the approach adopted by Langley and Cotoni (2004a). The TSEA equation
 141 that governs the ensemble average of the response of a system is given by Eq. (3). If a single
 142 realisation of the ensemble is considered then the mean TSEA equation is no longer applicable,
 143 but a similar equation can be written for the individual ensemble member that relates $d\mathbf{E}(t)/dt$,
 144 $\mathbf{E}(t)$ and $\mathbf{P}_{in}(t)$ following the same approach as Langley and Cotoni (2004a) and noting that
 145 the terms can be written as second order in the excitation. The SEA matrix, the power input,
 146 and the response energy will all differ from the ensemble average values, and the resulting
 147 equation for a single realisation can be written as

148
$$\frac{d}{dt} [\mathbf{E}(t) + \tilde{\mathbf{E}}(t)] + [\mathbf{A} + \tilde{\mathbf{A}}][\mathbf{E}(t) + \tilde{\mathbf{E}}(t)] = \mathbf{P}_{in}(t) + \tilde{\mathbf{P}}_{in}(t) \quad (4)$$

149 where a “tilde” over a variable represents the deviation from the ensemble average value. It
 150 has been shown by Langley and Cotoni (2004a) that power balance implies that the row sum
 151 of the SEA matrix must be zero, which implies (if N is the number of subsystems) that

152
$$\sum_j^N \tilde{A}_{jk} = 0 \quad \Rightarrow \quad \tilde{A}_{kk} = -\sum_{j \neq k}^N \tilde{A}_{jk}. \quad (5)$$

153 Hence the diagonal entries of the “random part” of the SEA matrix can be expressed in terms
 154 of the off-diagonal components. If Eq. (3) is subtracted from Eq. (4) then the result is

155
$$\frac{d\tilde{\mathbf{E}}(t)}{dt} + \mathbf{A}\tilde{\mathbf{E}}(t) + \tilde{\mathbf{A}}\mathbf{E}(t) + \tilde{\mathbf{A}}\tilde{\mathbf{E}}(t) = \tilde{\mathbf{P}}_{in}(t), \quad (6)$$

156 and this equation can be rewritten using the summation convention in the form

157
$$\frac{d\tilde{E}_j}{dt} + A_{jk}\tilde{E}_k + \tilde{A}_{jk}E_k + \tilde{A}_{jk}\tilde{E}_k = \tilde{P}_{in,j} \quad (7)$$

158 where the arguments of the various functions are omitted for ease of notation. Now Eq. (7)
 159 can be multiplied by \tilde{E}_r and averaged across the ensemble to yield

$$160 \quad \mathbb{E} \left[\tilde{E}_r \left(\frac{d\tilde{E}_j}{dt} + A_{jk}\tilde{E}_k + \tilde{A}_{jk}E_k + \tilde{A}_{jk}\tilde{E}_k - \tilde{P}_{in,j} \right) \right] = 0. \quad (8)$$

161 Likewise the r th component of Eq. (6) can be multiplied by \tilde{E}_j and averaged across the
 162 ensemble to yield

$$163 \quad \mathbb{E} \left[\tilde{E}_j \left(\frac{d\tilde{E}_r}{dt} + A_{rk}\tilde{E}_k + \tilde{A}_{rk}E_k + \tilde{A}_{rk}\tilde{E}_k - \tilde{P}_{in,r} \right) \right] = 0. \quad (9)$$

164 Equations (8) and (9) can be added, and on neglecting third order terms (or alternatively,
 165 assuming that these terms average to zero) this yields

$$166 \quad \dot{C}_{jr} = -A_{jk}C_{kr} - A_{rk}C_{kj} - E_k \left(q_r^{(jk)} + q_j^{(rk)} \right) + P_{jr} + P_{rj}, \quad (10)$$

167 where C_{kr} is the kr th component of the covariance matrix of the energies, and the following
 168 definitions are introduced

$$169 \quad C_{kr} = \mathbb{E}[\tilde{E}_k\tilde{E}_r], \quad q_r^{(jk)} = \mathbb{E}[\tilde{A}_{jk}\tilde{E}_r], \quad P_{jr} = \mathbb{E}[\tilde{P}_{in,j}\tilde{E}_r]. \quad (11 - 13)$$

170 Equation (10) is a matrix Riccati equation, with “forcing” arising from the final four terms on
 171 the right-hand side. Now a differential equation for the variables $q_r^{(jk)}$ can be derived by
 172 multiplying Eq. (7) by \tilde{A}_{nm} ($n \neq m$) and taking the ensemble average to yield

$$173 \quad \mathbb{E} \left[\tilde{A}_{nm} \left(\frac{d\tilde{E}_j}{dt} + A_{jk}\tilde{E}_k + \tilde{A}_{jk}E_k + \tilde{A}_{jk}\tilde{E}_k - \tilde{P}_{in,j} \right) \right] = 0. \quad (14)$$

174 By noting that the matrix \tilde{A}_{nm} does not depend upon time, Eq. (14) can be written in the form

$$175 \quad \dot{q}_j^{(nm)} = -A_{jk}q_k^{(nm)} - E_k \mathbb{E}[\tilde{A}_{jk}\tilde{A}_{nm}] + \mathbb{E}[\tilde{P}_{in,j}\tilde{A}_{nm}], \quad n \neq m. \quad (15)$$

176 Langley and Cotoni (2004a) have argued that, apart from the constraint imposed by Eq. (5),
 177 the entries of the matrix \tilde{A}_{nm} are uncorrelated, and moreover they are uncorrelated from the
 178 power inputs. Equation (15) therefore yields

$$179 \quad \dot{q}_j^{(nm)} = -A_{jk}q_k^{(nm)} - \delta_{jn}E_m \text{Var}[\tilde{A}_{nm}] + \delta_{jm}E_m \text{Var}[\tilde{A}_{nm}], \quad n \neq m \quad (16)$$

180 where $\text{Var}[\dots]$ represents the variance of its argument. Furthermore, the ‘‘diagonal’’ values of
 181 $q_j^{(nm)}$ can be expressed in terms of the off-diagonal entries by employing Eq. (5) to yield

$$182 \quad q_j^{(mm)} = - \sum_{n \neq m}^N q_j^{(nm)}. \quad (17)$$

183 A differential equation for the terms P_{jr} that appear in Eq. (10) can be obtained by multiplying
 184 Eq. (7) by \tilde{P}_r and taking the ensemble average to give

$$185 \quad \text{E} \left[\tilde{P}_{in,r} \left(\frac{d\tilde{E}_j}{dt} + A_{jk}\tilde{E}_k + \tilde{A}_{jk}E_k + \tilde{A}_{jk}\tilde{E}_k - \tilde{P}_{in,j} \right) \right] = 0. \quad (18)$$

186 Langley and Cotoni (2004a) have shown that the power input to different subsystems can be
 187 considered to be uncorrelated, which means that Eq. (18) can be written as

$$188 \quad \dot{P}_{rj} = -A_{jk}P_{rk} + \delta_{rj} \text{Var}[\tilde{P}_{in,j}]. \quad (19)$$

189 It is assumed here that $\tilde{P}_{in,r}$ is either constant or varies slowly with time such that $\dot{\tilde{P}}_{in,r}\tilde{E}_j \ll$
 190 $\tilde{P}_{in,r}\dot{\tilde{E}}_j$. Equations (10), (16), (17) and (19) form a set of first order differential equations that
 191 can be integrated to yield the variance of the response energy.

192 Note that P_{jr} arises from the presence of a *steady-state* power acting on the system, in addition
 193 to any shock loading applied at $t = 0$. If the system is subjected only to shock loading, then
 194 this can be applied as an initial condition on C_{kr} therefore all the terms relating to P_{jr} go to
 195 zero, and the differential equation for these terms, Eq. (19), is not needed. If the external power

196 input varies slowly with time relative to the oscillations of the system, accurate results are also
 197 expected. It can be shown (by removing time derivatives in Eqs. (10), (16) and (19) and noting
 198 that Eq. (10) can be split and solved as a collection of terms plus their transpose) that with a
 199 steady-state power input the solution to the transient variance equations is identical to the
 200 solution that is calculated using the steady-state theory of Langley and Cotoni (2004a).

201 The “Var” terms that appear in Eqs. (16) and (19) act as source terms, and they can be evaluated
 202 using the theory based on the Gaussian Orthogonal Ensemble described by Langley and Brown
 203 (2004a, 2004b). The relative variance, denoted $\text{RelVar}[\cdot]$, of the power inputs is given by

$$204 \quad \text{RelVar}[P_j] = \left(\frac{\alpha_j - 1}{\pi m_j B_j^2} \right) \{ 2B_j \tan^{-1} B_j - \ln(1 + B_j^2) \} + \frac{1}{(\pi m_j B_j)^2} \ln(1 + B_j^2). \quad (20)$$

205 Here m_j is the effective modal overlap factor of subsystem j , which is given by Langley and
 206 Cotoni (2004a) as

$$207 \quad m_j = \frac{1}{(\mathbf{C}^{-1})_{jj}}, \quad (21)$$

208 where \mathbf{C} is the conventional steady-state SEA matrix, $C_{jk} = A_{jk} n_k$. The parameter B_j in Eq.
 209 (20) is a bandwidth parameter defined by Langley and Cotoni (2004a) as

$$210 \quad B = \Delta \frac{n_j}{m_j}. \quad (22)$$

211 Finally, the parameter α_j depends on the nature of the applied loading, and for single point
 212 loading is found to be approximately 2.7 (Langley and Brown, 2004a).

213 The term $\text{Var}[\tilde{A}_{nm}]$ which appears in Eq. (16) can be evaluated by noting that for steady-state
 214 the relative variance of the matrix entry \tilde{A}_{nm} is given by Langley and Cotoni (2004a) as

$$215 \quad \text{RelVar}[\tilde{A}_{nm}] = \left(\frac{\alpha_{nm} - 1}{\pi m_n B_n^2} \right) \{ 2B_n \tan^{-1} B_n - \ln(1 + B_n^2) \} + \frac{1}{(\pi m_n B_n)^2} \ln(1 + B_n^2) \quad (23)$$

216 where values for the parameter α_{nm} are also given in Langley and Cotoni (2004a).

217 At early times, immediately after an impulse all modes of the system are in phase and so power
 218 transfer between subsystems is not a smooth function of time until the coherence between the
 219 modes is lost. If it is assumed over the ensemble that the natural frequencies take a uniform
 220 distribution, the instantaneous power transferred through the coupling between subsystems i
 221 and j can be written

$$\begin{aligned}
 222 \quad W_{ij}(t) &= \frac{1}{\Delta} \int_{\omega_1}^{\omega_2} W_{ij,av} (1 + \cos(2\omega t)) d\omega \\
 &= W_{ij,av} \left(1 + \cos((\omega_1 + \omega_2)t) \operatorname{sinc}(\Delta t) \right) \quad (24)
 \end{aligned}$$

223 where $W_{ij,av}$ is the average power transferred ω_1 and ω_2 are the lower and upper frequencies
 224 of the frequency band and $\Delta = \omega_2 - \omega_1$. Ignoring the fast frequency oscillations at the $\omega_1 +$
 225 ω_2 frequency, the variance of the power transfer and therefore the coupling loss factor matrix
 226 must be amended such that Eq. (23) becomes

$$\begin{aligned}
 227 \quad \operatorname{RelVar}[\tilde{A}_{nm}] &= (1 + \operatorname{sinc}(\Delta t))^2 \left\{ \left(\frac{\alpha_{nm} - 1}{\pi m_n B_n^2} \right) \{ 2B_n \tan^{-1} B_n - \ln(1 + B_n^2) \} \right. \\
 &\quad \left. + \frac{1}{(\pi m_n B_n)^2} \ln(1 + B_n^2) \right\}. \quad (25)
 \end{aligned}$$

228 Provided the bandwidth is reasonably wide, the effect of this modification, enacted by the sinc
 229 function, will decay after a short time.

230 By solving the mean TSEA equation, Eq. (3), the variance terms in the TSEA variance
 231 equations, Eqs. (16) and (19), can be calculated from the relative variance of the power input
 232 and coupling loss factors using Eqs. (20) and (25). All that remains for the TSEA variance
 233 equations to be integrated and solved is an appropriate set of initial conditions.

234

235

236 **A. The initial conditions**

237 The initial conditions on the C_{kr} , $q_r^{(jk)}$ and P_{jr} variables are case dependent; however under
 238 impulsive loading, there will be no external excitation so $P_{jr} = 0$ and the initial energy in a
 239 subsystem will be uncorrelated to its coupling loss factor so $q_r^{jk} = 0$. The initial condition on
 240 the variance of the subsystem energies will only be nonzero for impulsively excited subsystems
 241 and can be obtained by noting that the modal impulse response function (assuming unit
 242 generalized mass) is given by

$$243 \quad q_n(t) = \frac{\phi_n}{\omega_d} e^{-\beta\omega_n t} \sin(\omega_d t), \quad (26)$$

244 where β is the damping ratio, ω_n and ω_d are respectively the undamped and damped natural
 245 frequencies, and ϕ_n is the mode shape at the excitation point. The initial energy due to the
 246 impulse is therefore proportional to

$$247 \quad X = \sum_{\omega_n \in \Delta} \phi_n^2, \quad (27)$$

248 where the sum is over the number of modes in the frequency band. Equation (27) can also be
 249 written as

$$250 \quad X = \sum_n \phi_n^2 Y(\omega_n, \Delta), \quad (28)$$

251 where the function Y is zero unless the natural frequency falls in the band, in which case the
 252 function equals unity. The variance of the sum X can be obtained from random point process
 253 theory (Mehta, 2004) which yields the result

$$254 \quad \text{Var}[X] = E[\phi_n^4] \int_0^\infty Y^2(\omega_n, \Delta) f_1(\omega_n) d\omega_n + E[\phi_n^2]^2 \int_0^\infty \int_0^\infty Y(\omega_n, \Delta) Y(\omega_m, \Delta) g_2(\omega_n, \omega_m) d\omega_n d\omega_m, \quad (29)$$

255 where f_1 is equal to the modal density of the system, and $g_2(\omega_n, \omega_m)$ is a function known as
 256 the second cluster function. The double integral which appears in Eq. (29) is very complicated,
 257 but progress can be made by noting that: (i) the first integral yields simply N , the average
 258 number of modes in the band, and (ii) when $\phi_n = 1$ the equation yields the number variance,
 259 denoted $\text{Var}[N]$, which is the variance of the number of modes in the band. It can therefore be
 260 deduced that

$$261 \quad \text{Var}[X] = (E[\phi_n^4] - E[\phi_n^2]^2)N + E[\phi_n^2]^2 \text{Var}[N]. \quad (30)$$

262 The mean value of X is simply $E[\phi_n^2]N$, and so the relative variance is given by

$$263 \quad \text{RelVar}[X] = \frac{\alpha - 1}{N} + \frac{2}{(\pi N)^2} (\ln N + 2.18), \quad (31)$$

264 where the known result for the GOE number variance is employed (Mehta, 2004) and from
 265 (Langley and Brown, 2004a) $\alpha = E[\phi_n^4]/E[\phi_n^2]^2$ and is discussed below. The initial condition
 266 on the variance for any subsystem j that is subjected to an impulsive force, $f(t)$, is therefore

$$267 \quad E[\tilde{E}_j^2] = \left\{ \frac{\alpha_j - 1}{N_j} + \frac{2}{(\pi N_j)^2} [\ln N_j + 2.18] \right\} E_j^2(0), \quad (32)$$

268 where $E_j(0)$ is the initial mean energy and can be calculated using the method of Langley et
 269 al., 2019.

270 The result of Eq. (32) assumes that the square of the modeshapes, ϕ_n^2 , are uncorrelated from
 271 each other. However, depending on the specific application of the impulsive excitation, this
 272 correlation can be significant and act to reduce the initial energy variance. This poses a
 273 limitation on the applicability of Eq. (32) and is discussed further in Section VI.

274

275

276 **B. Selecting a suitable bandwidth**

277 Since any bandwidth of interest could be very wide, it can be split into smaller bandwidths, or
278 ‘sub-bands’, and the variance from each can be added together. This assumes that each sub-
279 band is sufficiently wide such that it is uncorrelated to neighbouring bands. The problem with
280 taking a very wide bandwidth is that it averages over the frequency detail of the response and
281 so can produce less accurate results. Sub-bands must therefore be selected that balance having
282 a wide enough bandwidth that each sub-band is uncorrelated and a narrow enough bandwidth
283 that significant frequency information is not lost. Empirical data for the modal statistics
284 suggests that $\alpha \approx 2.7$ (Langley and Brown, 2004a) and using the assumption of GOE statistics,
285 it has been suggested that interaction between approximately 18 neighbouring modes occurs
286 (Langley and Cotoni, 2004b) hence a bandwidth including 18 modes could be a sensible choice.
287 The effect of the sub-bandwidths is illustrated in the numerical validation of Section IV. Within
288 each sub-band a centre frequency must be selected and it has been found that it is important to
289 use the frequency at which the steady-state response is equal to the band-averaged response.
290 This provides improved results over simply selecting the mean frequency of the sub-band.

291

292 **IV. Numerical validation**

293 In this section, the method of Section III for predicting the variance of the subsystem energies
294 in an SEA system is validated via comparison with results calculated using the finite element
295 method (FE). A finite element package is used to model plate structures and generate the
296 stiffness and mass matrices required for subsequent modal analysis.

297 **A. Methodology**

298 Since SEA is generally applied to energy at high frequencies, any impulse applied can be
299 thought to only excite within a given frequency band. Here this is applied as an impulse that

300 excites only the modes in the frequency range of interest. The contribution of each of these
301 modes is calculated for a given impulse location and the modal impulse responses are
302 superimposed to generate the total impulse response of the system. For a narrow frequency
303 range, this can provide poor results immediately after the impulse for subsystems that are not
304 directly forced since there are insufficient modes in the band to adequately cancel when
305 summed and produce zero energy far from the input.

306 In order to generate an ensemble of structures to compare to TSEA, the plates are randomised
307 by adding to each plate ten masses each of 2% of the plate mass and two springs of stiffness
308 $(250 \times 2\pi)^2 M_j$ at random locations where M_j is the mass of the j th plate. These values are
309 selected to generate sufficient statistical overlap in the subsystem natural frequencies to
310 produce GOE statistics as discussed by Kessissoglou and Lucas, 2009. The location of each
311 mass and spring is random, although it is ensured that they are not placed too close to the force
312 location.

313 The plate parameters used are for 5 mm thick Aluminium plate with Young's modulus 70 GPa,
314 density 2700 kg m⁻³, Poisson ratio 0.33, loss factor 0.01 and with all edges pinned to allow
315 rotation, but zero displacement. Only out-of-plane modes are considered and an ensemble of
316 1000 realisations over a frequency range of 500-1500 Hz is used and is considered as a single
317 frequency band unless otherwise stated. The FE solution uses second order triangular shell
318 elements with a minimum element length of less than one sixth of the wavelength of vibration
319 at the highest frequency of interest. The TSEA solution was performed with a timestep of
320 2×10^{-4} s.

321

322

323

324 **B. Prediction of the variance of the subsystem energies**

325 For the system shown inset in Figure 1a, the mean and standard deviation TSEA results for
326 plates one and two are compared to FE results in Figure 1 where plate one is subjected to an
327 impulse. The system consists of two plates of areas 1.26 and 1.3 m² coupled via three point
328 connectors that are constrained to only move in the vertical direction and behave like 1D
329 springs of stiffness 2.5×10^6 N m⁻¹. The standard deviation is plotted instead of the variance
330 so the results can be more easily compared to the mean. The force location is varied with each
331 realisation and the CLFs are calculated analytically by modelling the connectors as springs. In
332 order to ensure that any errors in the transient variance results are due to deficiencies in the
333 method rather than errors carried over from the mean TSEA or steady-state SEA variance
334 methods, the mean TSEA results are tuned to fit the FE by modifying the CLFs and the steady-
335 state band-averaged variance from SEA is also tuned to closely match the variance from the
336 FE simulations by modifying the variance parameters.

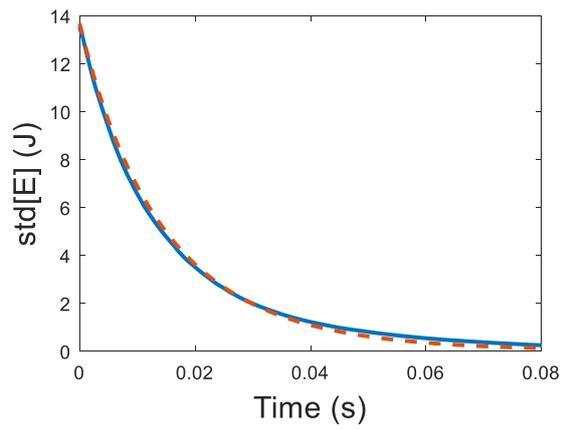
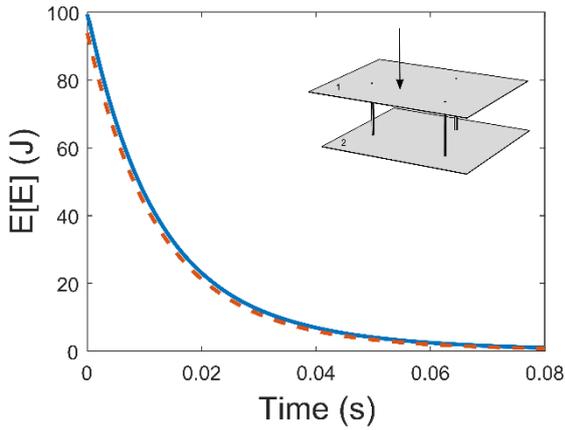
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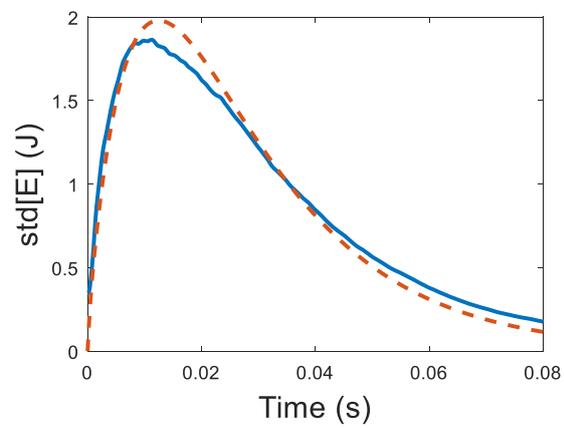
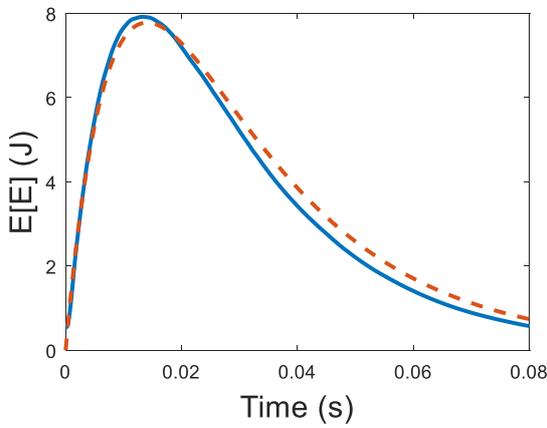
Time (s)

Time (s)

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Time (s)

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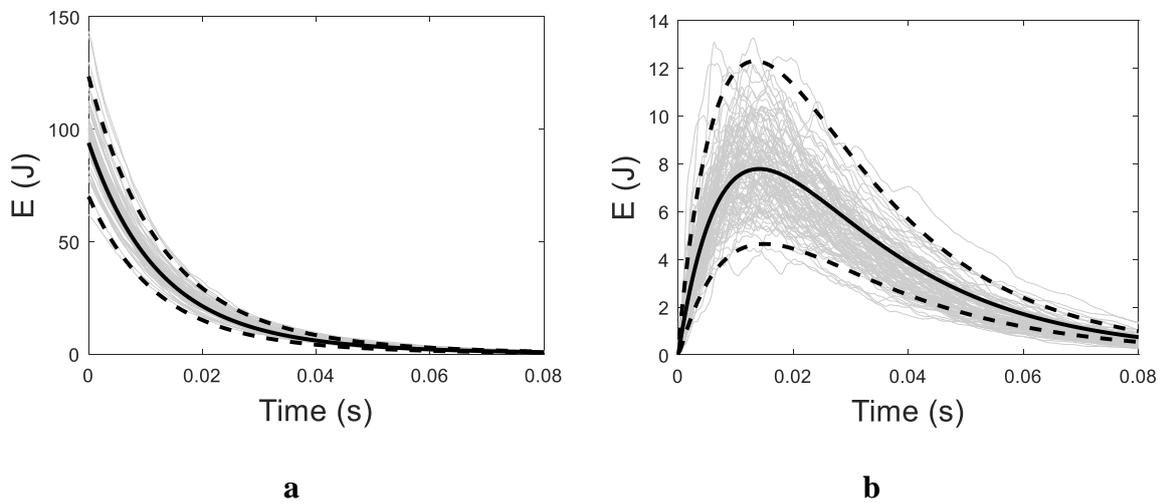
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346 **Figure 1: (Colour online) TSEA results (dashed) compared with FE results (solid) for a)**
 347 **plate one mean, b) plate one standard deviation, c) plate two mean and d) plate two**
 348 **standard deviation.**

349

350 The standard deviation results show good agreement with the FE results. The variance of the
 351 initial energy in plate one is well predicted by Eq. (32) and the decay is also matched well. In
 352 plate two, the rise rate and time of maximum variance is very well predicted, with the TSEA
 353 results slightly overpredicting the maximum variance by approximately 7% and yielding a
 354 slightly slower peak time. The decay in the variance is close, but not perfectly matched to the
 355 FE results and this can be a consequence of band-averaging as discussed later.

356 To illustrate the necessity of calculating the variance as well as the mean of the ensemble,
 357 Figure 2 presents the 95% confidence intervals for both plates, with 100 realisations of the
 358 response superimposed. The variance provides an accurate measure of the spread of the
 359 response and the subsystem energies of any one realisation could be under or over predicted if
 360 only the mean value is used. It is assumed here that the probability density function of the
 361 energy of each subsystem exhibits a lognormal distribution as discussed by Langley et al.
 362 (2013). At any one time five realisations should lie outside the 95% confidence interval and
 363 this is observed to be approximately true.

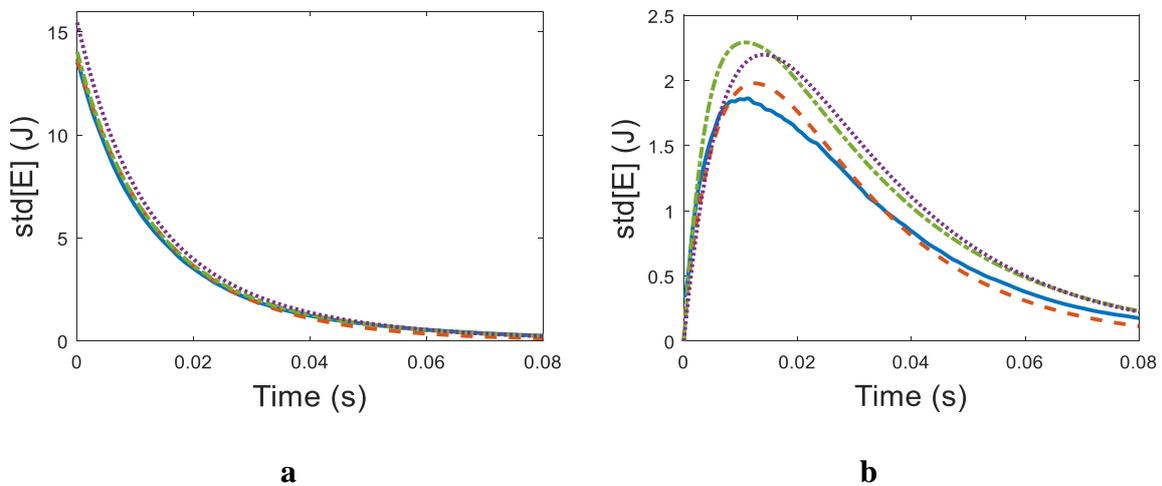


366 **Figure 2: 95% confidence interval (black dotted) from the TSEA results (mean in solid**
 367 **black) compared with 100 realisations (grey), a) plate one and b) plate two.**

368

369 The effect of using different sub-bands is investigated in Figure 3 where the standard deviation
 370 for both plates is plotted over the overall frequency range 500-1500 Hz with 1, 10 and 100 sub-
 371 bands. The average modal density of the two plates is 0.082 modes/Hz meaning that there are
 372 approximately 8 modes per 100 Hz in each subsystem. The sub-bandwidth has a noticeable
 373 effect on the standard deviation, in particular the decay rate, initial conditions, peak values and
 374 peak times. The initial condition in plate one is overestimated with the smallest sub-bands

375 because the assumption that bands are uncorrelated breaks down. Whilst the predictions of
 376 plate two appear worse for smaller sub-bands, this is likely due to the steady-state band-
 377 averaged variance being tuned for the case of a single sub-band and improving the tuning for
 378 the narrower sub-bands should improve accuracy. Additionally, as the sub-bandwidth reduces,
 379 the decay rate approaches that of the FE results illustrating that accuracy can decrease with a
 380 wide bandwidth.

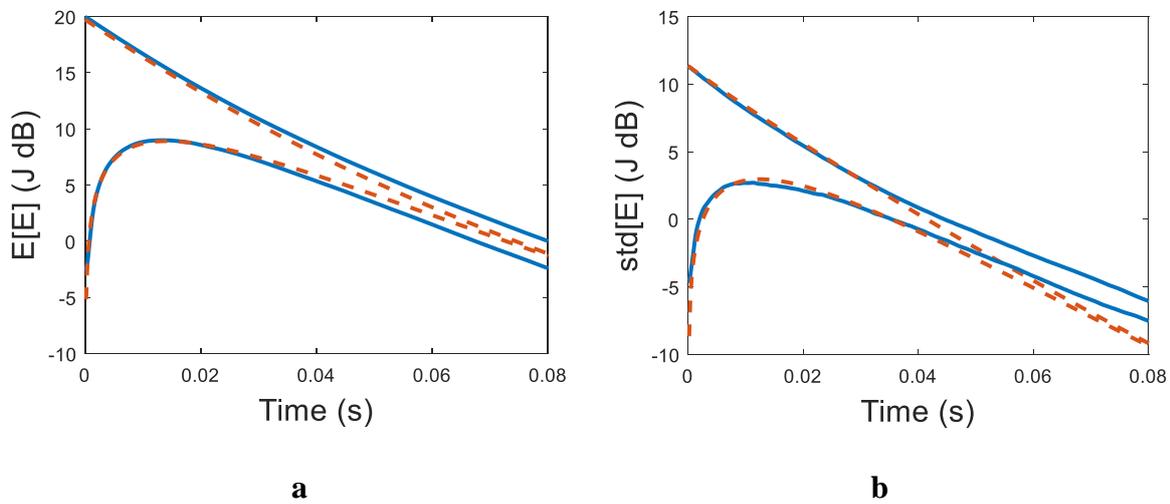


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Figure 3: (Colour online) Standard deviation of the energy from FE (solid) and TSEA with 1 (dashed), 10 (dash-dot) and 100 (dotted) sub-bands for a) plate one and b) plate two.

387 It is important to discuss what level of accuracy the TSEA variance approach can be expected
 388 to provide and what is considered sufficient when applied in real engineering design scenarios.
 389 Due to similar assumptions in the derivation, it is reasonable to expect similar accuracy from
 390 transient SEA as is achieved in steady-state SEA and therefore the two should be compared. It
 391 is generally accepted that errors of ± 3 dB (or approximately a factor of two) are reasonable for
 392 steady-state SEA especially when considering the relative difference between subsystem
 393 energies can be orders of magnitude. The predictions of the peak energy standard deviation

394 observed here are within 1 dB of the benchmark results. Additionally, It is most common to
 395 plot SEA results on a logarithmic scale, whereas Figures 1-3 are displayed on a linear scale.
 396 The results of Figure 1 are replotted on a logarithmic scale in Figure 4 where any errors now
 397 look relatively small when considering the large drop in the variance of the energy between the
 398 plates is well predicted. The decay rate of the variance of the energy in both plates is not
 399 predicted particularly well by the TSEA and this is thought to be a feature of the band-averaging
 400 as discussed above.



403 **Figure 4: (Colour online) a) Mean and b) standard deviation of the energy from FE**
 404 **(solid) and TSEA (dashed) plotted on a logarithmic scale. Plates one and two are the top**
 405 **and bottom lines respectively.**

406

407 The variance equations are also applicable when a steady-state load is applied. For the case
 408 where the system used in Figure 1 is initially at rest and plate one is given a stationary band-
 409 limited load with constant spectrum in the frequency range 500-1500Hz, the mean and variance
 410 FE and TSEA results are shown in Figure 5. The load is applied as white noise that only acts
 411 on the modes within the frequency band and the equations of motion are reformulated into the
 412 non-stationary Lyapunov equation to be solved with numerical integration. In these figures,

413 horizontal lines are included to represent the results that are obtained from a purely steady-state
414 analysis and it is clear that the TSEA mean and variance results asymptote towards these values,
415 illustrating that the TSEA equations collapse to the steady-state equations under stationary
416 conditions. As before, the SEA parameters such as the CLF and input power are tuned to
417 provide close agreement with the FE results so that the transient response from the TSEA
418 method can be reasonably compared. For both the mean and standard deviation, close
419 agreement between the predicted TSEA and benchmark FE results is observed, although the
420 rise time of the standard deviation in plate one is slightly overpredicted.

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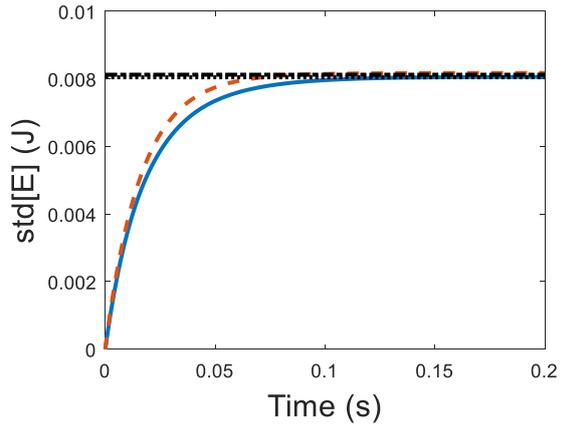
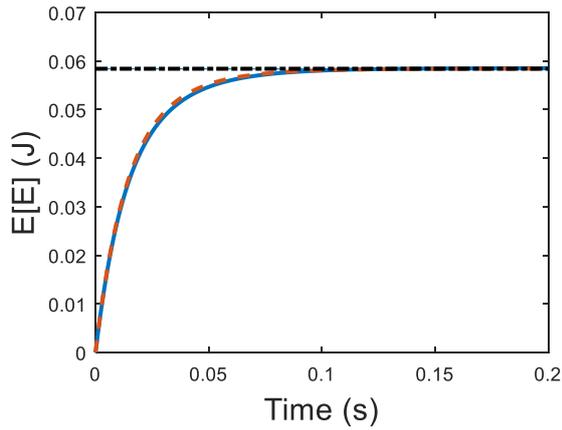
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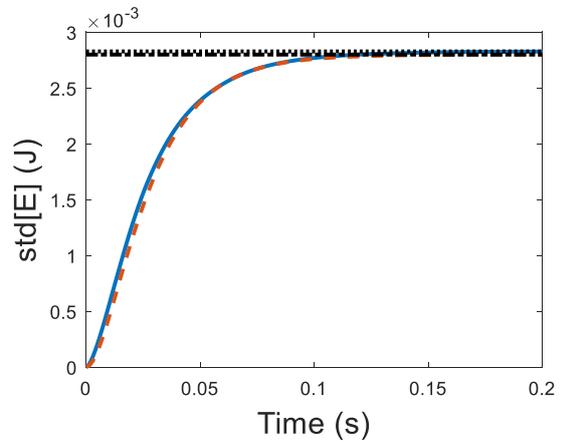
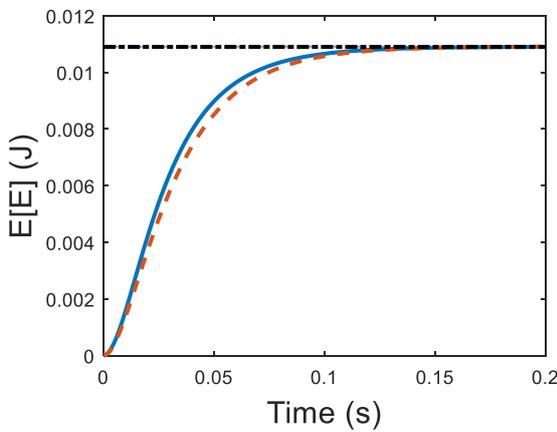


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434 **Figure 5: (Colour online) TSEA results (dashed) compared with FE results (solid) for**
 435 **plate one of the two plate system under steady loading. a) plate one mean, b) plate one**
 436 **standard deviation, c) plate two mean and d) plate two standard deviation. Dash-dot**
 437 **and dotted lines represent results from steady-state FE and SEA respectively.**

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444 **V. Experimental validation**

445 To supplement the numerical validation of the TSEA method in Section IV, an experiment
446 involving a two-plate system similar to that displayed in Figure 1 is also presented. Two
447 aluminium plates with area 0.8 and 0.48 m² are suspended using string and coupled with two
448 steel point connections as displayed inset in Figure 6a. A sharp impulse with high-frequency
449 content is applied with a stiff-tipped impulse hammer containing a force transducer at an
450 approximately consistent location to the upper plate, denoted plate one, and the responses of
451 the plates are measured using five randomly spaced accelerometers on each plate from which
452 the average velocity and therefore an estimate of the energy of each plate is calculated. In order
453 to randomise the modeshapes and generate an ensemble of systems, a number of masses,
454 totalling approximately 10% of the plate mass, are attached to each plate and are redistributed
455 for each impulsive excitation.

456 The loss factors of each subsystem are determined at a number of frequencies by investigating
457 the decay curves of the subsystems in isolation and are found to be approximately 0.01. The
458 SEA coupling loss factors are then calculated from the experimental steady-state energy
459 difference between each plate using

$$460 \quad \eta_{12} = \frac{n_2 E_2}{n_2 E_1 - n_1 E_2} \eta_2 \quad (33)$$

461 where the modal densities are calculated analytically for a plate and η_{21} can be calculated using
462 the reciprocity relationship $n_1 \eta_{12} = n_2 \eta_{21}$. The modal overlap factors at 1000 Hz are 1.25 and
463 0.75 for plates one and two respectively.

464 A slight extension to the variance theory of Section III must be included to account for the
465 effect of estimating each subsystem energy from a finite number of points. This increases the
466 observed energy variance since, in addition to the variance across the ensemble given by TSEA,

467 denoted $\text{Var}_{\text{ens}}(E_j)$, there is a spatial variance, denoted $\text{Var}_{\text{spa}}(E_j)$, due to spatial fluctuations
468 in energy over the subsystem at any one time. The total variance measured from the experiment
469 is therefore $\text{Var}_{\text{ens}}(E_j) + \text{Var}_{\text{spa}}(E_j)$. If there are N observation points and a realisation with
470 average energy \bar{E}_j has energy at the n th point given by $E_{j,n} = \bar{E}_j + \tilde{E}_{j,n}$ then the spatial variance
471 is given by $E[\tilde{E}^2]/N$. The spatial relative variance for a system with Gaussian modes is
472 unity and so the spatial variance becomes \bar{E}^2/N .

473 The experimental results are compared in Figure 6 to transient mean and variance results in the
474 frequency range 2000-3000 Hz where 30 realisations are taken, and the data is normalised such
475 that the peak force of each impulse is 1 N. The similarity between the two is very strong, with
476 comparable rise and decay times, although the TSEA slightly underestimates the peak standard
477 deviation in plate two. Similar to the mean results, oscillations are observed in the standard
478 deviation and would be reduced by a larger ensemble or more accelerometers as discussed in
479 Langley et al., 2019.

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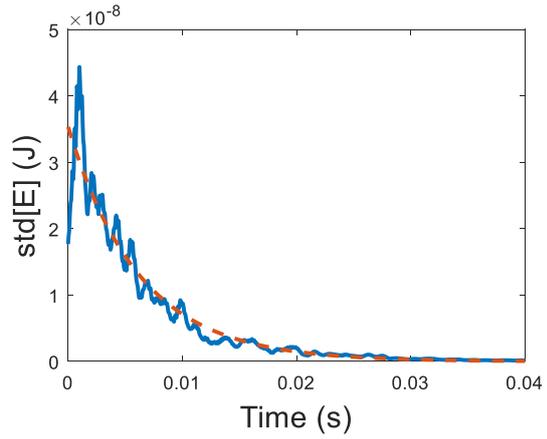
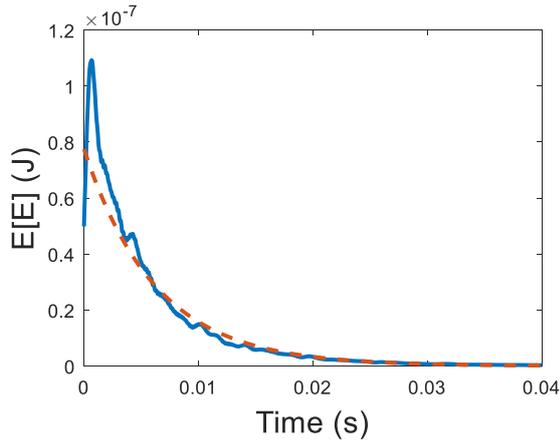
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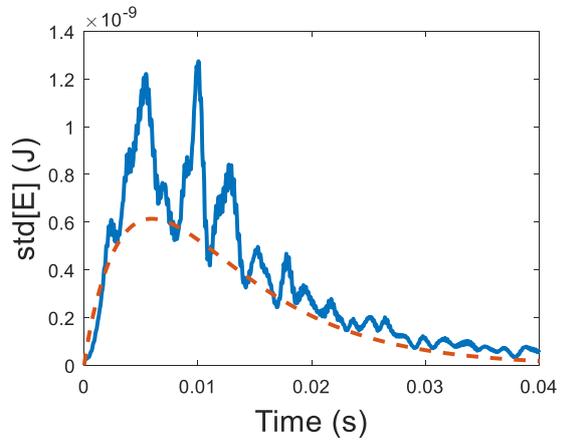
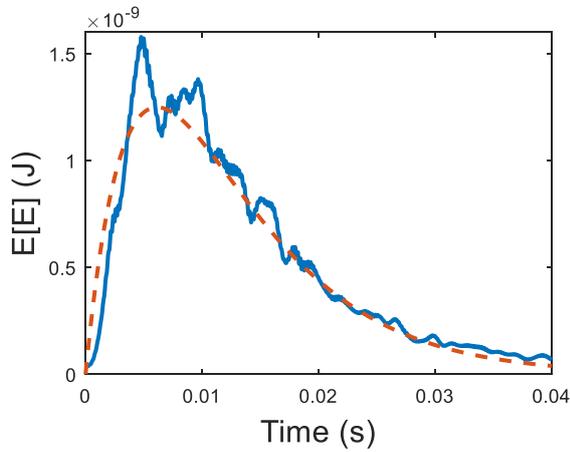


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490 **Figure 6: (Colour online) TSEA results (dashed) compared with experimental results**
 491 **(solid) for a) plate one mean, b) plate one standard deviation, c) plate two mean and d)**
 492 **plate two standard deviation.**

493

494 VI. Limitations of the method

495 For the cases presented in Sections IV and V, the variance predictions from the derived method
 496 exhibit strong agreement with the benchmark results. However, the method is limited by the
 497 assumptions it makes thus for certain systems and loading conditions, the predictions are less
 498 accurate. These limitations are explored in this section.

499 The results of Figure 1 are obtained by varying the impulse location with each realisation. In
500 this case, the variance of the initial energy in the excited plate is reasonably predicted by Eq.
501 (32), although in fact this will generally be an overprediction since it ignores the effect of any
502 correlations between modeshapes. In the arguably more realistic case where the impulse
503 location is fixed throughout the ensemble, the effects of the correlations become greater, acting
504 to reduce the initial variance and so Eq. (32) overpredicts by an amount dependent on the
505 bandwidth of interest. This effect is displayed in Figure 7 where the impulse location is fixed
506 and the overprediction can be seen in both plates and the impact of selecting the initial variance
507 to match the FE results is shown by the dash-dot curve, which displays good accuracy. It might
508 be suggested that the correlations can be predicted by the GOE correlations (Brody et al., 1981),
509 although this has been found to be an overestimate. Should an accurate model of these
510 modeshape correlations be derived, the derived theory is expected to exhibit strong accuracy
511 as suggested by the dash-dot curve in Figure 7.

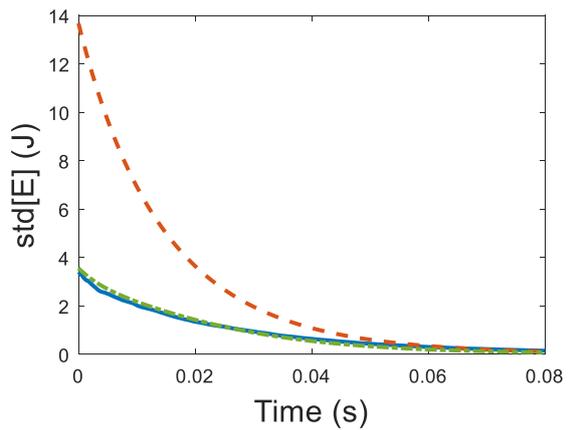
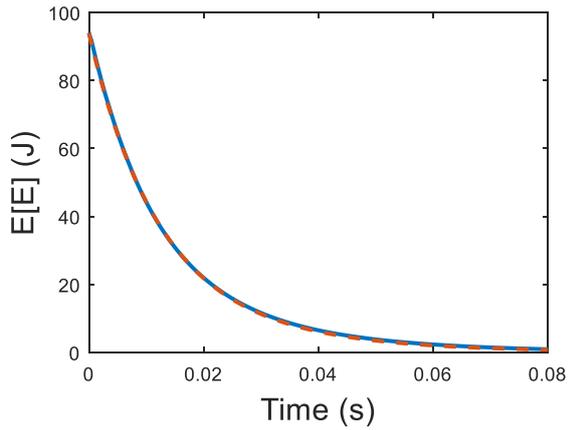
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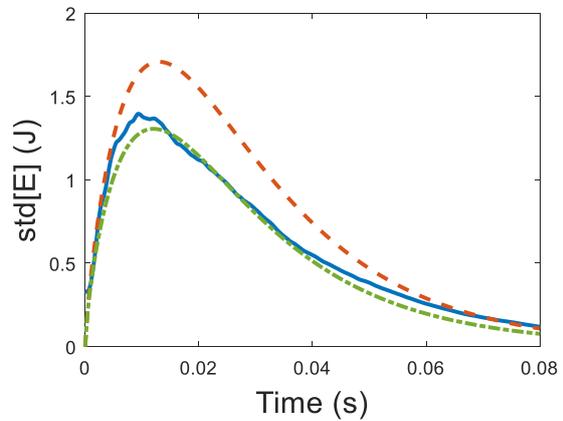
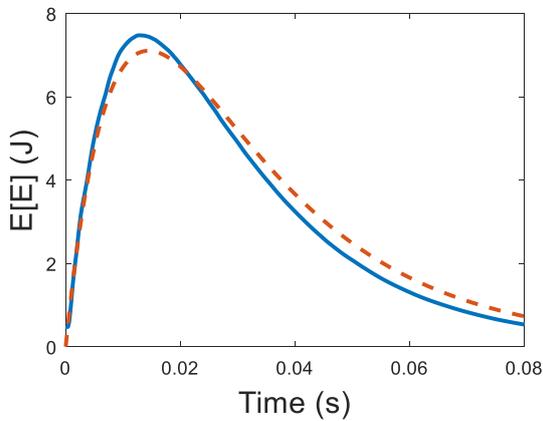


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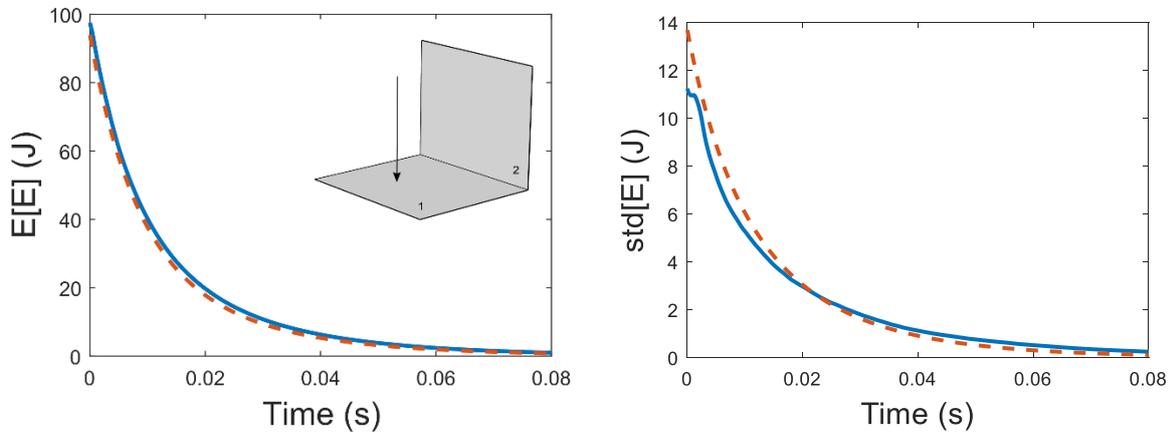
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521 **Figure 7: (Colour online) TSEA results (dashed) compared with FE results (solid) for**
 522 **the point coupled system when the impulse location is constant. The dash-dot curve**
 523 **displays the TSEA results when the initial variance in plate one is selected to match the**
 524 **initial variance found in the FE results. a) plate one mean, b) plate one standard**
 525 **deviation, c) plate two mean and d) plate two standard deviation.**

526

527 A second limitation to the method is illustrated clearly by the results from the two edge-coupled
 528 plates displayed in Figure 8 where the impulse location is varied randomly with each
 529 realisation. A clear early peak in the standard deviation in the second plate is observed in the
 530 FE results in Figure 8d, but not the theoretical results. Physically, this is due to waves spreading

531 out from the different impulse locations of each realisation taking different amounts of time to
 532 reach the edge coupling and start transferring energy through to the second subsystem.

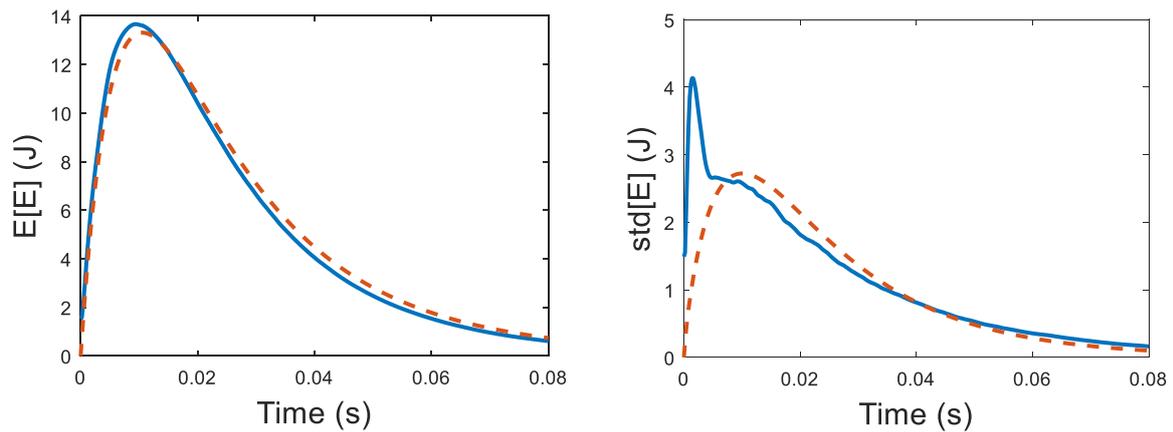


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537 **Figure 8: (Colour online) TSEA results (dashed) compared with FE results (solid) for**
 538 **the edge-coupled system with a varying impulse location. a) plate one mean, b) plate one**
 539 **standard deviation, c) plate two mean and d) plate two standard deviation.**

540

541 The discrepancy between the theory and simulations in Figure 8d is due to the theory being
 542 unable to account for the actual method of energy transfer. In Eq. (16), the variance from the
 543 coupling between plates assumes a diffuse wavefield in both plates when in reality, shortly
 544 after an impulse, the wavefield is a propagating wavefield and a wavefront can approach the

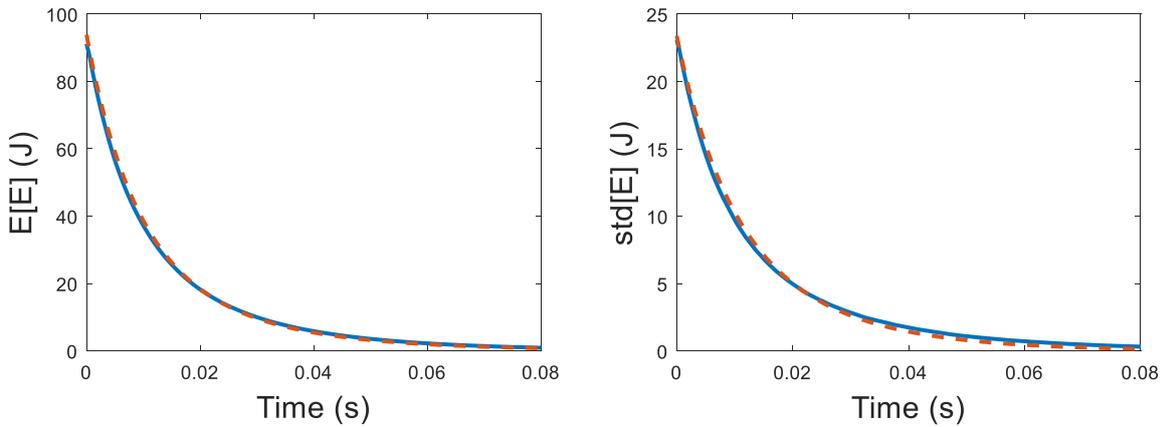
545 junction and cause high instantaneous transmission. It is interesting to note that this effect
546 averages out and the mean energy is unaffected. In this case, the variance prediction at later
547 times may be unreliable due to cumulative effects from the poor prediction at early times.

548 The accuracy of the method can be estimated by comparing the timescales involved in energy
549 transport. If the timescale for a subsystem to achieve a diffuse wavefield after an impulse is
550 short compared to the timescale of energy transfer between subsystems, then the derived TSEA
551 variance method can be expected to give a good prediction. This occurs when the coupling is
552 weak and explains the strong accuracy of the results of Figure 1 where the coupling is weaker.
553 Additionally, if the damping is low then energy transfer between subsystems occurs over a
554 longer time and so is greater, since the energy decay in the excited subsystem will take longer.
555 Any effect from the non-diffuse field at early times will therefore be less significant.

556 Weak coupling and low damping are well-known criteria for the reasonable application of
557 steady-state SEA. In fact, it has been suggested that a measure of coupling strength and validity
558 of SEA can be taken from mean time-domain energy plots such as Figure 8c by taking the ratio
559 of peak time to total time duration (Fahy, 1996; James and Fahy, 1997). A value of coupling
560 strength indicator of 0.07 is suggested as a reasonable threshold above which the coupling can
561 be considered weak in an SEA sense. From Figure 8c, the edge coupled system is found to
562 have a coupling strength indicator of 0.15, well above the threshold, although the variance still
563 displays a discrepancy with theory. It is therefore suggested that for reasonable application of
564 the TSEA variance method, weaker coupling is required than for steady-state SEA. It should
565 be noted that despite this limitation, the accuracy of the method is still very strong and well
566 within the expected accuracy of SEA-based approaches.

567 To investigate the effect of the wave field in the impulsively excited plate, fifty simultaneous
568 impulses are applied at random locations with zero mean and unity standard deviation Gaussian

569 random magnitudes with the intention of instantaneously generating a more diffuse field in the
 570 plate. The results are presented in Figure 9 where the strong peak in the standard deviation of
 571 the second plate is removed and the TSEA variance method provides accurate predictions. This
 572 reinforces the above discussion suggesting that the method is limited by the diffuse field
 573 assumption. In this case, the initial energy variance in the excited plate is matched to the FE
 574 results rather than found analytically.

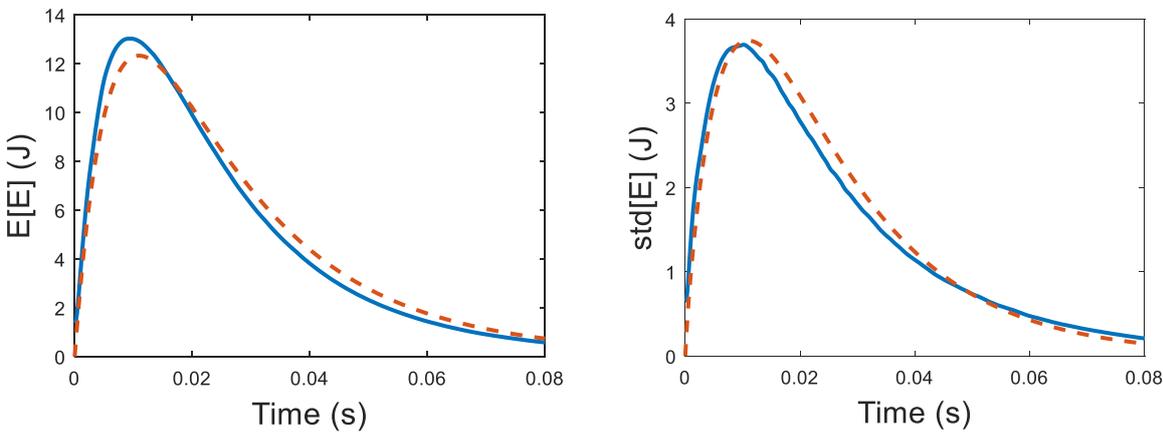


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579 **Figure 9: (Colour online) TSEA results (dashed) compared with FE results (solid) for**
 580 **the edge coupled system with 50 simultaneous random impulses. a) plate one mean, b)**
 581 **plate one standard deviation, c) plate two mean and d) plate two standard deviation.**

582

583 **VII. Conclusions**

584 The transient SEA method is extended to predict the variance of the energy in each subsystem
585 of a built-up system under both impulsive and steady loads. Prediction of both the mean and
586 the variance of a system response is important since at high frequencies, nominally identical
587 structures can behave differently, and some measure of this variability is important when
588 designing a structure.

589 Following a similar procedure to Langley and Cotoni (2004a), a power balance equation for
590 each individual realisation of a system is manipulated to generate a differential equation for the
591 covariance of the subsystem energies, Eq. (10). Since the variance in the response arises from
592 variability in coupling loss factors and input power, the covariance equation is ‘forced’ by
593 terms related to these variance quantities, Eqs. (16) and (19). The variance forcing terms are
594 themselves calculated from differential equations where any variance terms are found from the
595 ensemble statistics of the system (Langley and Brown, 2004a, 2004b). When impulsive
596 excitation is applied, the variability across the ensemble of the energy injected into the system
597 is imposed as an initial condition on the energy covariance matrix. Under steady-state loading,
598 the method is shown to collapse to the steady-state SEA variance method of Langley and
599 Cotoni (2004a).

600 The applicability and limitations of the derived method are investigated both numerically and
601 experimentally and it is found to provide close agreement with benchmark results in cases of
602 weak coupling. However, when coupling is stronger, the theoretical predictions can become
603 less accurate due to the assumption of a diffuse field occurring instantaneously in an
604 impulsively excited subsystem becoming less reasonable. Additionally, the variance of the
605 initial energy in a subsystem that provides the initial condition for the covariance equation
606 cannot yet be calculated accurately when a constant impulse location is used due to correlations

607 between modeshapes. This is a complex issue and outside of the scope of this paper, but it
608 should be noted that if the force location is not held constant, the correlations become less
609 significant and a reasonable initial condition can be applied. Despite these limitations, strong
610 results are observed and errors generally lie well within the accuracy expected from SEA-based
611 methods.

612

613 **Acknowledgements**

614 The authors would like to thank Mitsubishi Heavy Industries for funding this research.

615

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673 **List of figure captions**

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675 plate one standard deviation, c) plate two mean and d) plate two standard deviation.

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677 Figure 2: 95% confidence interval (black dotted) from the TSEA results (mean in solid black)
678 compared with 100 realisations (grey), a) plate one and b) plate two.

679

680 Figure 3: Standard deviation of the energy from FE (solid) and TSEA with 1 (dashed), 10
681 (dash-dot) and 100 (dotted) sub-bands for a) plate one and b) plate two.

682

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684 (dashed) plotted on a logarithmic scale. Plates one and two are the top and bottom lines
685 respectively.

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688 plate system under steady loading. a) plate one mean, b) plate one standard deviation, c) plate
689 two mean and d) plate two standard deviation. Dash-dot and dotted lines represent results
690 from steady-state FE and SEA respectively.

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697 TSEA results when the initial variance in plate one is selected to match the initial variance

698 found in the FE results. a) plate one mean, b) plate one standard deviation, c) plate two mean
699 and d) plate two standard deviation.

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701 Figure 8: TSEA results (dashed) compared with FE results (solid) for the edge-coupled
702 system with a varying impulse location. a) plate one mean, b) plate one standard deviation, c)
703 plate two mean and d) plate two standard deviation.

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705 Figure 9: TSEA results (dashed) compared with FE results (solid) for the edge coupled
706 system with 50 simultaneous random impulses. a) plate one mean, b) plate one standard
707 deviation, c) plate two mean and d) plate two standard deviation.