

# Flame Double Input Describing Function analysis

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## Abstract

The Flame Describing Function (FDF) is a useful and relatively cheap approximation of a flame's nonlinearity with respect to harmonic velocity fluctuations. When embedded into a linear acoustic network, it is able to predict the amplitude and stability of harmonic thermoacoustic oscillations through the harmonic balance procedure. However, situations exist in which these oscillations are not periodic, but their spectrum contains peaks at several incommensurate frequencies. If one assumes that two frequencies dominate the spectrum, these oscillations are quasiperiodic, and the FDF concept can be extended by forcing the flame with two amplitudes and two frequencies. The nonlinearity is then approximated by a Flame Double Input Describing Function (FDIDF), which is a more expensive object to calculate than the FDF, but contains more information about the nonlinear response.

In this study, we present the calculation of a non-static flame's FDIDF. We use a  $G$ -equation-based laminar conical flame. We embed the FDIDF into a thermoacoustic network and we predict the nature and amplitude of thermoacoustic oscillations through the harmonic balance method. A criterion for the stability of these oscillations is outlined. We compare our results with a classical FDF analysis and self-excited time domain simulations of the same system. We show how the FDIDF improves the stability prediction provided by the FDF. At a numerical cost roughly equivalent to that of two FDFs, the FDIDF is capable to predict the onset of Neimark-Sacker bifurcations and to identify the frequency of oscillations around unstable limit cycles. At a higher cost, it can also saturate in amplitude these oscillations and predict the amplitude and stability of quasiperiodic oscillations.

*Keywords:* Flame Describing Function, Premixed flame response, Thermoacoustic oscillations

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## 1. Introduction

Thermoacoustic oscillations are a persistent problem in rocket and gas turbine engines. While their onset can be modelled with linear methods, prediction of their finite amplitude behaviour requires the use of nonlinear techniques. In the last decade these

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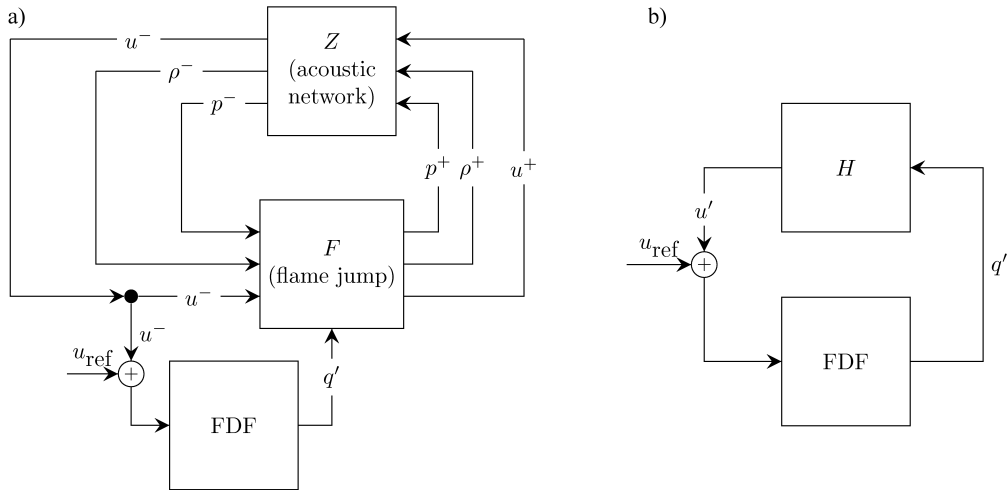


Figure 1: Overview of a closed-loop thermoacoustic network.  $\rho$ ,  $p$  and  $u$  denote the flow density and acoustic pressure and velocity variables. (a) The jump across the flame element is highlighted, using superscripts  $-$  and  $+$  for the acoustic properties upstream and downstream the flame, respectively. All the remaining acoustic information is embedded into the acoustic block  $Z$ . (b) The same closed-loop thermoacoustic network simplified across the FDF element;  $H$  contains all the linear acoustic response with respect to heat release perturbations.

5 nonlinear methods have involved both frequency domain [1, 2, 3, 4, 5, 6] and time domain  
6 methods [7, 8, 9, 10, 11, 12].

7 Time domain methods tend to be computationally expensive. One usually converts  
8 the (linear) frequency response of a given acoustic system into the time domain by using  
9 Green's functions [7, 9, 10], Fourier modes [5, 11] or a state space approach [8, 13, 14].  
10 One then couples this with a nonlinear flame model and performs a simulation forward  
11 in time, exploiting the full nonlinear characteristics of the flame model. However, this is  
12 usually expensive, even for low-order models, because many oscillation cycles have to be  
13 simulated before the final attractor of the system is reached [11]. Numerical continuation  
14 algorithms [12] are cheaper, but require smooth numerical integration techniques. They  
15 can predict limit cycle oscillations and their stability, but not the amplitude of non-  
16 periodic oscillations.

17 On the other hand, frequency domain methods tend to be cheap. Rather than sim-  
18 ulate the entire system's nonlinear behaviour, one encapsulates the flame's nonlinear  
19 response with a Flame Describing Function (FDF), which is the frequency response of  
20 the flame with respect to harmonic forcing at variable amplitude. FDF methods were  
21 introduced in thermoacoustics by [15], and first fully exploited by [6]. The calculation of  
22 an FDF can be expensive, but if the flame model is not changed, the same FDF can be  
23 used to test many acoustic configurations with a low-cost procedure, known as harmonic  
24 balance.

25 For the harmonic balance analysis, the nonlinear flame dynamics is decoupled from  
26 the acoustics. This can be done if the flame is acoustically compact, meaning that the  
27 characteristic flame length  $L_f$  is much shorter than the smallest acoustic wavelength of

28 interest  $\lambda_{\min} = c/f_{\max}$ . Under this assumption, a generic thermoacoustic configuration  
 29 can be drawn as a block diagram as in Fig. 1a. The acoustic jump conditions across  
 30 the flame have been highlighted. Their inputs are the acoustic variables upstream of  
 31 the flame and the instantaneous heat release fluctuations,  $q'$ . The acoustics is expressed  
 32 in terms of downstream ( $f$ ) and upstream ( $g$ ) travelling waves. The remaining acoustic  
 33 response is contained in the upstream and downstream acoustic blocks. For the simple  
 34 configuration composed of two straight ducts interconnected by a flame, they contain  
 35 information about the mean flow, end reflection coefficients, and wave time delays [3].  
 36 Finally, the FDF converts velocity disturbances upstream of the flame into heat release  
 37 fluctuations.

38 The feedback loop in Fig. 1a can be simplified by choosing a reference input signal  
 39  $u_{\text{ref}}$  just upstream of the FDF and the heat release as an output, so that the open-loop  
 40 heat release response with respect to velocity fluctuations is given by:

$$q' = \text{FDF}(u' + u_{\text{ref}}) \quad (1)$$

41 Furthermore, the entire open-loop acoustic response with respect to heat release fluctua-  
 42 tions can be embedded into a unique transfer function  $H$  so that, for velocity fluctuations  
 43 upstream of the flame, we can write (see Fig. 1b):

$$u' = Hq' \quad (2)$$

44 For simple acoustic networks, the expression for the transfer function  $H$  can be found  
 45 analytically [3, 14]. It becomes rather complicated for complex networks, and numerical  
 46 methods are used in these cases to evaluate  $H$  over a certain range of frequencies.

Closing the feedback loop between the velocity at the reference point and the heat  
 release fluctuations yields:

$$q' = \frac{\text{FDF}(A, s)}{1 - \text{FDF}(A, s)H(s)}u_{\text{ref}} \quad (3)$$

47 Equation (3) represents a Single Input Single Output system: if no input velocity is  
 48 prescribed, the system will be linearly unstable if and only if it has poles in the r.h.s. of  
 49 the complex plane in the zero amplitude limit. Looking for these poles is equivalent to  
 50 finding solutions to the harmonic balance dispersion relation

$$\text{FDF}(A, s)H(s) = 1 \quad (4)$$

51 where  $A$  is the upstream velocity amplitude and  $s = \sigma + i\omega$  is the Laplace variable. The  
 52 dispersion relation (4) is also able to identify poles which have a negative growth rate at  
 53 small amplitudes, but become unstable at finite amplitudes. This is a characteristic of  
 54 subcritical Hopf bifurcations, and phenomena such as bistability and triggering may be  
 55 observed.

56 Solving the dispersion relation (4) at various amplitudes leads to harmonic limit cycle  
 57 solutions of the closed-loop system, for which the growth rate  $\sigma$  is equal to zero. Their  
 58 stability may be analysed by investigating the change in growth rate across the saturated  
 59 amplitude [6, 14, 16]. These solutions are, however, only harmonic approximations of  
 60 the actual response of the system. Furthermore, if the growth rate of more than one  
 61 thermoacoustic mode is positive, then the oscillations are non-periodic. Because the

62 FDF is calculated by forcing the flame harmonically, it cannot be used to predict the  
63 amplitude of non-periodic oscillations. In particular, one cannot linearly superpose two  
64 periodic solutions that are found from the harmonic balance at a given operating point.  
65 This is simply because the flame’s behaviour when forced by two finite amplitude signals  
66 is not a linear superposition of its behaviour when forced by each finite amplitude signal  
67 independently. When using the FDF, therefore, one cannot rule out the possibility that  
68 the long time behaviour is non-periodic.

69 A detailed investigation of the interplay between two oscillating modes has been per-  
70 formed experimentally only for simple configurations [17], due to its high cost. Nonethe-  
71 less, the presence of multiple, incommensurate frequencies in the spectrum of thermoac-  
72oustic oscillations has been reported in several experimental studies [18, 19, 20]. The  
73 study of the nonlinear interaction between the modes may be relevant for the analysis of  
74 these systems. It has also been observed in experiments that, although a single eigenmode  
75 is found to be linearly unstable, nonlinear effects may actually stabilise the oscillations at  
76 this frequency and trigger oscillations at a different frequency [6]. The FDF can predict  
77 the existence of oscillations of the two modes independently, but will fail in predicting  
78 their stability, as the latter is connected to the nonlinear coupling between the two modes.  
79 This phenomenon is usually called mode-switching, and was observed also by [21] and in  
80 gas turbines experiments by [22, 23]. In [21] it was shown that mode-switching can be  
81 attributed to the existence of an unstable quasiperiodic attractor in the phase-space of  
82 thermoacoustic trajectories, which the FDF framework cannot calculate.

83 In order to predict the amplitude of at least some classes of non-periodic oscillations,  
84 a different approximation of the nonlinear flame model has to be calculated. This is  
85 known as the Double Input Describing Function (DIDF), and is created by forcing the  
86 flame with a signal composed of two harmonic components with independent amplitudes  
87 and incommensurate frequencies [24]. The calculated Flame DIDF (FDIDF) can then be  
88 fed into an acoustic network in a similar manner to that in Fig. 1. The harmonic balance  
89 procedure yields two coupled dispersion relations which have to be solved simultaneously,  
90 as was first shown by [21] for a thermoacoustic system.

91 The aim of this study is to present a numerical analysis that exploits frequency  
92 domain calculations of a non-static (or dynamic<sup>1</sup>) nonlinearity based on a low-order  
93 model for the flame dynamics. This is the major difference between our analysis and  
94 that of [21], where a static model for the flame was considered. For static nonlinearities,  
95 a Wiener-Hammerstein model can be adopted, which decouples the nonlinear amplitude  
96 saturation process from the linear dynamic response. This is not possible for dynamic  
97 nonlinearities, and the FDIDF we calculate is a nonlinear object that couples the input  
98 amplitudes and frequencies. We also obtain an analytical criterion for the stability of  
99 quasiperiodic oscillations, which is different from the one discussed in [21]. A different  
100 attempt to extend the concept of the FDF was proposed by [25, 26], where higher order  
101 transfer functions that account for modal coupling were derived using Volterra series  
102 expansions. However, the dependence of the higher order transfer functions upon the  
103 relative amplitude of the input modes was not considered in these studies. The non-static  
104 model we adopt for the flame is the kinematic nonlinear  $G$ -equation, which is known to  
105 lead to quasiperiodic oscillations when coupled with an acoustic network [11, 12, 14].

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<sup>1</sup> A nonlinearity is non-static if it depends on time derivatives of the input state.

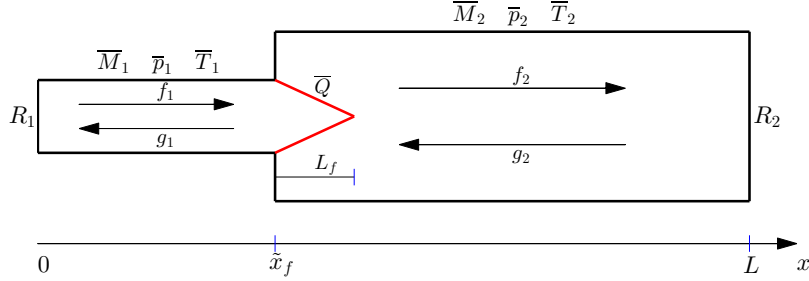


Figure 2: Sketch of the thermoacoustic network. A compact flame connects two ducts with different diameters and mean temperatures. The linearised Euler equations are solved on top of a uniform mean flow solution. Jump conditions at the flame and prescribed reflection coefficients at the inlet/outlet provide closure for the model.

106 The FDIDF method is able to predict the location of Neimark-Sacker bifurcations, the  
 107 frequency of unstable oscillations around limit cycles, and also the saturation amplitude  
 108 and the stability of quasiperiodic oscillations. The study is structured as follows: in §2  
 109 we describe the acoustic and flame configurations we investigate; in §3 we present FDF  
 110 results and calculate harmonic limit cycles amplitudes and frequencies, together with  
 111 their stability, highlighting strong points and weaknesses of the method; in §4 and §5 the  
 112 FDIDF is presented and tested against the FDF in the limit of a small forcing amplitude;  
 113 the dispersion relations which couple it with the acoustic response are derived and solved;  
 114 the frequencies and amplitudes of periodic and non-periodic solutions are calculated with  
 115 the harmonic balance method based on the FDIDF; a criterion for the stability of these  
 116 solutions is outlined; results are compared with the FDF method analysis and with time  
 117 domain simulations of the same nonlinear system. Finally in §6 the study is summarised  
 118 and the benefits and problems of the methods are discussed.

## 119 2. Thermoacoustic model

120 The thermoacoustic model we will consider throughout this study consists of a lam-  
 121 inar, conical flame confined in a simple acoustic network. The same model has been  
 122 presented and extensively discussed in [14, 27] but is summarized here for completeness.

### 123 2.1. Acoustic network

124 The acoustic network we consider is shown in Fig. 2. It consists of two intercon-  
 125 nected ducts with different cross sectional areas. A flame, assumed to be acoustically  
 126 compact, is located just after the area change. Rankine-Hugoniot jump conditions for  
 127 mass, momentum and energy are solved across the area change and the flame to guaran-  
 128 tee the conservation of these fluxes [3]. Reflection coefficients, which may be frequency  
 129 dependent, are specified at the inlet and outlet. This is a simple but generic model for a  
 130 combustion driven Rijke tube, such as the one analysed experimentally by [28].

131 The geometry and mean flow parameters are: total length of the combustor  $L =$   
 132 860 mm; upstream and downstream duct diameters  $D_1 = 23$  mm and  $D_2 = 25.6$  mm;  
 133 inlet temperature  $\bar{T}_1 = 300$  K; Mach number upstream of the flame  $\bar{M}_1 = 0.0057$ ; outlet

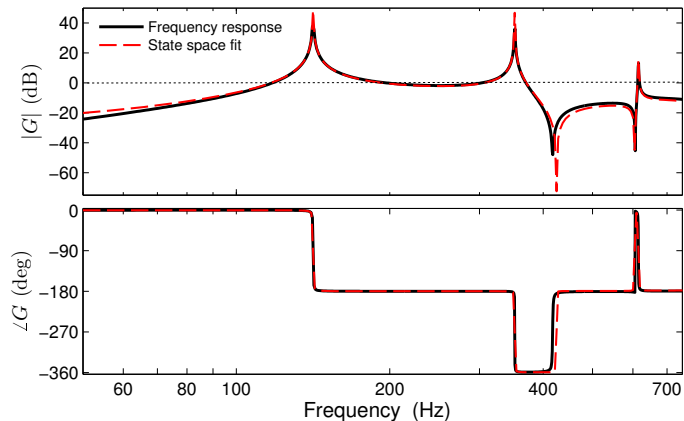


Figure 3: Comparison between the frequency response calculated with LOTAN (solid black) and the state space approximation (dashed red) at  $x_f = 0.34$ . The approximation works well over a wide range of frequencies.

134 pressure equals to atmospheric pressure; temperature ratio across the flame  $\bar{T}_2/\bar{T}_1 =$   
 135 2. Note that the average temperature in the downstream duct,  $\bar{T}_2$ , is lower than the  
 136 adiabatic flame temperature in order to account for heat losses through the walls. We  
 137 will use the non-dimensional position of the flame in the duct, defined as  $x_f \equiv \tilde{x}_f/L$ ,  
 138 where  $\tilde{x}_f$  is the dimensional flame position value, as a bifurcation parameter. Finally, we  
 139 use a frequency dependent reflection coefficients at the inlet and outlet when solving the  
 140 acoustic equations. We choose the low-Mach number limit of the reflection coefficient  
 141 derived analytically by [29, 30] using the Wiener-Hopf technique. It has been validated  
 142 against experiments in [31].

143 Using the Low-Order ThermoAcoustic Network (LOTAN) framework [4, 32], the  
 144 acoustic network is solved in the frequency domain by decomposing the acoustic vari-  
 145 ables into upstream and downstream travelling waves (see Fig. 2). We calculate the  
 146 acoustic eigenfrequencies and, by imposing harmonic fluctuations in the heat release at  
 147 the combustion zone, we evaluate the open-loop acoustic transfer function  $H_{x_f}$  as in  
 148 eq. (2). Using both the frequency response and the eigenfrequencies, the acoustic re-  
 149 sponse to heat release fluctuations can be fitted onto a state space, as described in [14].  
 150 This is necessary to extend the frequency response – calculated at  $s = i\omega$  – in the full  
 151 Laplace space, in which the growth rate  $\sigma$  can be non-zero. Fig. 3 shows that the state  
 152 space approximation fits well the frequency response evaluated with LOTAN over a wide  
 153 range of frequencies.

154 Note that, by moving the flame position, the acoustic response of the system will  
 155 change, but the flame response will not. Thus, a different acoustic transfer function (and  
 156 its state space approximation) has to be evaluated each time the bifurcation parameter  
 157  $x_f$  is changed. This, however, is a cheap calculation. On the other hand, only one  
 158 (expensive) Flame Describing Function calculation has to be performed on the flame.  
 159 The same FDF can be used to study the stability of the system for any value of the  
 160 bifurcation parameter.

161 *2.2. Flame model*

162 As a model for the flame, we use the nonlinear kinematic  $G$ -equation to track the  
 163 flame front, which is located at the  $G=0$  level set [33, 34, 35]. We consider a laminar,  
 164 conical, axisymmetric flame. The flow field is assumed to be incompressible and, to  
 165 simplify the calculations, we neglect the density jump across the flame. The local flame  
 166 speed depends on the local flame curvature  $\kappa$  through the relation  $s_L = s_L^0(1 - \mathcal{L}\kappa)$ ,  
 167 where  $\mathcal{L}$  is the Markstein length and  $s_L^0$  the speed of a flat, laminar flame. The flow field  
 168 is composed of a uniform mean axial velocity  $\bar{U}$ , on top of which forced perturbations are  
 169 imposed, denoted with primes and described below. The perturbations are specified at  
 170 the burner inlet  $x_b$  and then travel at a characteristic velocity  $K$  in the flame domain [36].  
 171 We fix the value of the convective speed to  $K = 1.2\bar{U}$ , which is within the range obtained  
 172 numerically by [37] for a laminar conical flame. Under these assumptions, the  $G$ -equation  
 173 model is:

$$\frac{\partial G}{\partial t} + u'_r \frac{\partial G}{\partial r} + (\bar{U} + u') \frac{\partial G}{\partial x} = s_L^0(1 - \kappa\mathcal{L}) \sqrt{\left(\frac{\partial G}{\partial r}\right)^2 + \left(\frac{\partial G}{\partial x}\right)^2} \quad (5)$$

We solve this equation with the efficient Narrow Band Level Set method technique [38, 39]. We choose an incompressible travelling wave as a model for the perturbation flow [36, 37, 40], which reads:

$$\begin{aligned} \frac{\partial u'}{\partial t} + K \frac{\partial u'}{\partial x} &= 0 & u'(x = x_b) &= u'_{ac}(t) \\ \frac{1}{r} \frac{\partial (ru'_r)}{\partial r} + \frac{\partial u'}{\partial x} &= 0 & u'_r(r = 0) &= 0 \end{aligned} \quad (6)$$

174 Here  $u'_{ac}$  denotes the acoustic perturbation imposed at the inlet. The total heat release  
 175 is then given by

$$Q = \int_{G=0} \rho s_L^0 h_r (1 - \mathcal{L}\kappa) |\nabla G| r \, dr \, dx \quad (7)$$

176 where  $\rho$  is the flow density and  $h_r$  the heat released per unit mass. This  $G$ -equation  
 177 based model has been extensively studied both in the linear [27, 41, 42, 43, 44, 45] and  
 178 nonlinear [12, 14, 35, 37, 46] regimes when the imposed perturbations are harmonic,  
 179 i.e.  $u'_{ac} = A \sin(\omega t)$ . The main goal of this paper is to extend the nonlinear analysis in  
 180 the frequency domain to the case in which the inlet perturbation is given by the sum  
 181 of two incommensurate harmonic fluctuations, resulting in quasiperiodic oscillations.  
 182 Nonetheless, the harmonic case is instructive and its discussion is needed to present  
 183 some of the assumptions we will use in the quasiperiodic analysis and to benchmark the  
 184 FDIDF calculations.

185 **3. FDF analysis**

186 By FDF, we refer to the frequency domain approximation of the nonlinear flame re-  
 187 sponse to harmonic velocity perturbations (see Figure 4). We therefore set  $u'_{ac} = A \sin(\omega t)$ ,  
 188 time march eq.(5)-(6), and calculate the heat release according to eq. (7). Given that the

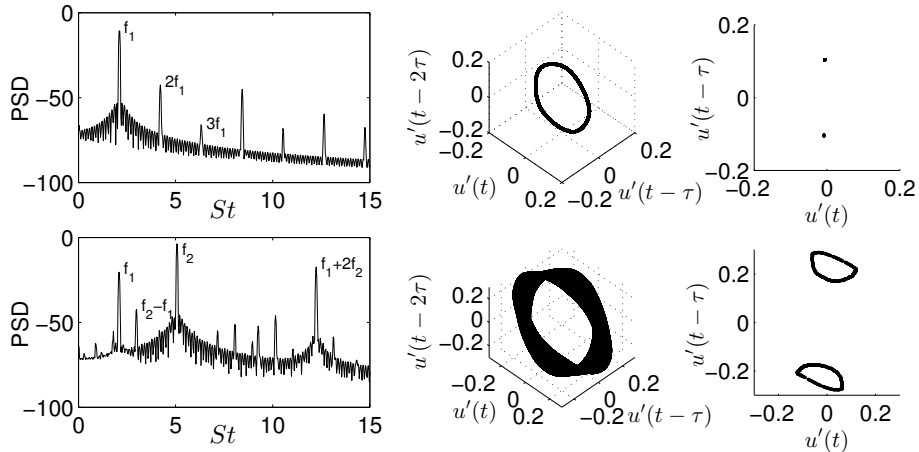


Figure 4: Velocity Power Spectral Density (PSD), phase plane and Poincaré sections of self-excited thermoacoustic oscillations. Top row: periodic oscillations. The system responds also at the harmonics, which are neglected in the FDF framework. Bottom row: quasiperiodic oscillations. The system responds also at the harmonics and linear combination of the fundamental frequencies, which are neglected by the FDIDF. Image reproduced from [14] with permission from Cambridge University Press.

189 velocity perturbation is harmonic with angular frequency  $\omega$ , it is reasonable to assume  
 190 that the heat release response can be expanded in a Fourier series as:

$$Q = \sum_{k=1}^{\infty} \hat{q}_k \sin(k\omega t + \phi_k) \quad (8)$$

191 This assumes that the heat release is periodic, with the same period as the forcing.  
 192 For laminar flames, this is supported by experimental evidence [42, 43, 47, 48]. This  
 193 model cannot capture a possible response of the nonlinearity at subharmonics. Also,  
 194 for laminar flames that oscillate in the absence of forcing at an intrinsic frequency [49],  
 195 it cannot capture the response that may appear at non-integer multiples of the forcing  
 196 frequency.

197 The FDF that is fed into the dispersion relation (4) is then defined as

$$\text{FDF}(A, i\omega) \equiv \frac{\hat{q}_1 e^{i\phi_1} \bar{U}}{\hat{u} \bar{Q}} \quad (9)$$

198 where  $\hat{u}$  is the Fourier component of the input velocity signal at the burner.

199 Rather than performing the FDF calculations over all possible frequencies, in the  
 200 following we provide an argument that allows us to limit the calculations only over  
 201 certain sets of dangerous frequencies. We first recall that the dispersion relation (4) can  
 202 be derived from the harmonic balance method [24]. Its solutions, which for a fixed value  
 203 of the amplitude can be interpreted as the poles of the closed-loop thermoacoustic system,  
 204 are those for which the loop-gain,  $|\text{FDF}||H|$ , is equal to 1 and the total (wrapped) phase  
 205 is equal to 0. To find limit cycle oscillations, we impose the additional condition that the  
 206 growth rate is equal to zero. From the loop-gain condition, one can infer that a necessary



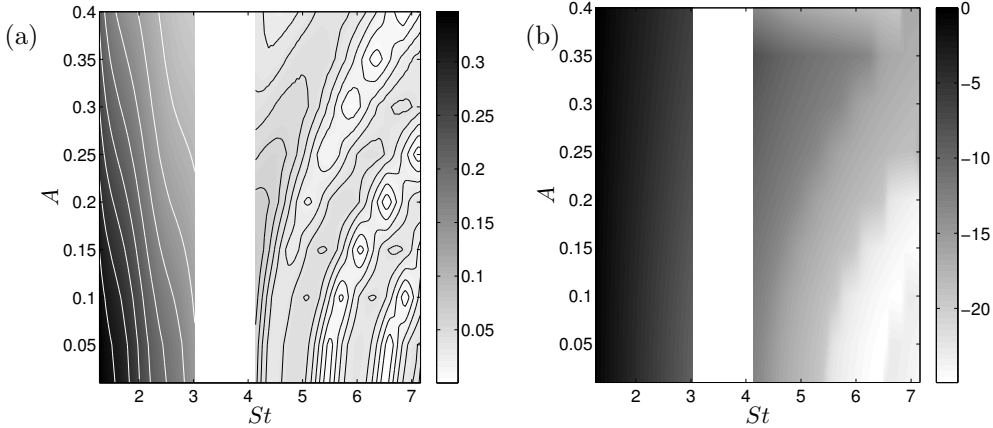


Figure 5: Gain (a) and phase (b) of a laminar, conical flame's FDF. The FDF is evaluated numerically only around frequencies which may give rise to thermoacoustic oscillations.

207 (but not sufficient) condition for a thermoacoustic oscillation to exist is that either the  
 208 acoustic transfer function  $H$  or the FDF must have a gain larger than 1. For the acoustic  
 209 transfer function, this happens close to the acoustic eigenfrequencies, whereas the FDF  
 210 may or may not have regions in which the gain is larger than 1. If the FDF gain is larger  
 211 than 1 over some frequency range, the frequency of thermoacoustic oscillations may lie  
 212 in this region, and can be far from the acoustic eigenfrequencies [50]. It has been shown  
 213 that these oscillations may persist even in the extreme case in which anechoic boundary  
 214 conditions for the acoustic network are imposed, and therefore no purely acoustic mode  
 215 exists [51, 52, 53, 54]. For this reason, these thermoacoustic modes have been labelled  
 216 as intrinsic thermoacoustic modes.

217 When the unconfined laminar conical flame model we are considering is forced har-  
 218 monically, its gain  $|\text{FDF}|$  never exceeds 1. Within the  $G$ -equation framework, this can  
 219 be proven analytically in the low forcing amplitude limit when curvature corrections on  
 220 the flame speed are neglected [27, 41, 45]. Numerical and experimental studies show  
 221 that this holds true even in the fully nonlinear case [14, 35, 48, 55]. For this reason, no  
 222 intrinsic thermoacoustic instabilities can be observed in our system, and we can deduce  
 223 that thermoacoustic oscillations are possible only in certain frequency bands, given by  
 224 the regions in which the acoustic gain  $|H|$  is larger than one. For example, at  $x_f = 0.34$   
 225 one can see from Fig. 3 that oscillations can be expected only in the [118, 197] Hz and  
 226 [303, 371] Hz band regions.

227 This is useful information because we can reduce the cost of the FDF calculations  
 228 by evaluating the FDF only over these frequency regions<sup>2</sup>. We identify these regions  
 229 while varying the bifurcation parameter  $x_f$  over the entire range [0 1]. Let us define the  
 230 Strouhal number  $St \equiv L_f f / \bar{U}$ , where  $L_f$  and  $\bar{U}$  are the characteristic flame length and

<sup>2</sup>A broad knowledge of the FDF is needed to ensure that a flame's gain never exceeds unity. For our model, we already have this information from the literature [14].

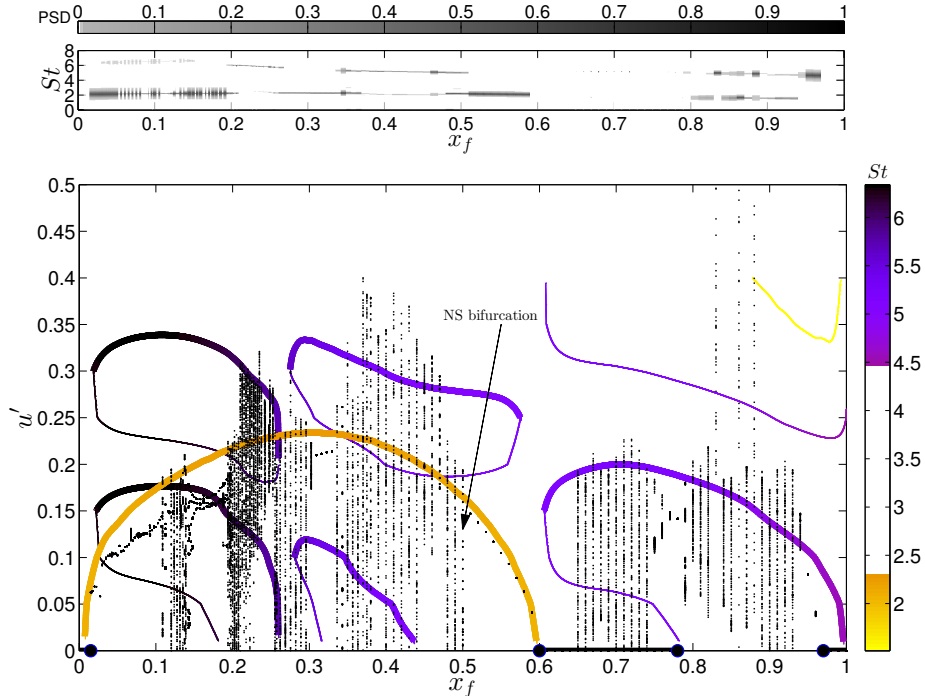


Figure 6: Bottom frame: bifurcation diagrams calculated with time domain simulations (dots) and the FDF method (lines), with the flame position,  $x_f$ , as the bifurcation parameter. Thick and thin lines correspond to stable and unstable limit cycles, respectively, and their colour to the limit cycle oscillation frequency. Dots represent peaks of time domain simulations, as described in [14]. The FDF method predicts Hopf bifurcations and periodic oscillations well (e.g. between  $x_f = 0.50$  and  $0.60$ ) but cannot predict quasiperiodic oscillations (e.g. between  $x_f = 0.34$  and  $0.50$ ). The top graph (reproduced from [14] with permission from Cambridge University Press) shows the PSD of time domain simulations at every flame location.

231 mean flow speed respectively. For the thermoacoustic system under consideration in this  
 232 study, oscillations are possible only in the frequency ranges  $St \in [1.273, 3.024]$ , associ-  
 233 ated with the fundamental acoustic eigenfrequency, and  $St \in [4.138, 7.162]$ , associ-  
 234 ated with the second acoustic eigenfrequency. Note that this range is obtained by considering  
 235 all possible values of  $x_f$  and is therefore different from the one discussed in the previous  
 236 paragraph, because the latter was considering only a specific position of the flame. We  
 237 carry out a detailed evaluation of the FDF in these frequency ranges, varying the ampli-  
 238 tude of the oscillation between 0 and 0.4. The FDF gain is shown in Fig. 5, and contains  
 239 the usual features of conical, premixed flames: the gain is larger at low frequencies and  
 240 overall it tends to decrease with the amplitude, a signature of the nonlinearity saturation  
 241 effect. This holds true at low frequencies, whereas at high frequencies the gain can also  
 242 increase with the amplitude, meaning that subcritical bifurcations and triggering may  
 243 be observed.

244 Having calculated both the acoustic transfer function  $H$  and the FDF, we can close  
 245 the thermoacoustic feedback loop as in Fig. 1 and calculate the thermoacoustic eigenfre-

246 quencies according to the dispersion relation (4). We recall that (4), deriving from the  
 247 harmonic balance method, works well when the so-called filtering hypothesis is satisfied,  
 248 meaning that the closed-loop system does not respond greatly at the harmonics of the  
 249 input frequency. Experimental and numerical studies have shown that laminar, conical  
 250 flames act as low-pass filters [41, 43, 55], and Fig. 5 shows that our model contains this  
 251 feature. Also the acoustics tends to damp high frequencies more, although gain peaks are  
 252 found at the resonance frequencies. For these reasons, we shall assume that the filtering  
 253 hypothesis is satisfied.

254 Limit cycles are found when the dispersion relation (4) is satisfied, with the additional  
 255 constraint that the growth rate of the oscillations is equal to zero. At this stage, no  
 256 further approximation has been introduced, because we have knowledge of the FDF at  
 257 harmonic oscillations. However, to assess the stability of the cycles we need to perturb  
 258 the saturation amplitude and calculate the shift in frequency and growth rate that it  
 259 causes, i.e., we want to find the  $\Delta s = \Delta\sigma + i\Delta\omega$  that satisfies

$$\text{FDF}(A_{LC} + \Delta A, s_{LC} + \Delta s) H(s_{LC} + \Delta s) - 1 = 0 \quad (10)$$

260 where  $A_{LC}$  and  $s_{LC} = i\omega_{LC}$  are a limit cycle solution of (4), and  $\Delta A$  is an imposed  
 261 infinitesimal perturbation. If  $\Delta\sigma/\Delta A$  is positive, the limit cycle is unstable, and if it is  
 262 negative then the cycle is stable. To solve (10) the FDF needs to be extended into the  
 263 complex plane  $\mathbb{C}$ . Following [56], we have tried two different techniques: (i) extrusion,  
 264 by assuming that the FDF does not vary with  $\sigma$ , and (ii) analytical continuation, by  
 265 fitting every amplitude slice of the FDF onto a state space. Both methods give the same  
 266 results.

267 The bifurcation diagram we obtain by varying the flame position is shown in Fig. 6.  
 268 Thick and thin lines indicate stable and unstable limit cycles as predicted by the har-  
 269 monic balance method. Results are compared with time domain simulations of the same  
 270 system [14]. Solid lines at  $A = 0$  indicate regions in which the time domain simulations  
 271 are linearly stable, and Hopf bifurcations are marked with circles. The dots represent  
 272 peaks of velocity fluctuations in the time domain. At  $x_f$  locations where multiple dots  
 273 are plotted, the oscillations therefore are non-periodic. For example, at  $x_f = 0.5$  time  
 274 domain oscillations cease to be periodic, and quasiperiodic solutions arise through a  
 275 Neimark-Sacker bifurcation, marked with an arrow in Fig. 6. The two methods give  
 276 similar locations of Hopf bifurcations and amplitudes of periodic oscillations. However,  
 277 the FDF method fails to predict the amplitude of non-periodic oscillations. Further,  
 278 although many stable limit cycles are predicted by the FDF method, time domain sim-  
 279 ulations rarely converge to these solutions. This is because they are not, in fact, stable.  
 280 The FDF criterion for stability misses this because it only considers growth or decay of  
 281 the mode that is already oscillating. It cannot consider growth or decay of another mode  
 282 on top of the oscillating limit cycle, which is considered with the FDIDF in the next  
 283 section.

284 A similar comparison between the time and frequency domain methods can be found  
 285 in [14]. However, in that study a mismatch between the locations of the Hopf bifurcations  
 286 predicted by the FDF and those found by time integration was observed. In this study we  
 287 have resolved the FDF more accurately in the frequency range in which thermoacoustic  
 288 oscillations are expected, which has led to a better match between the two methods.

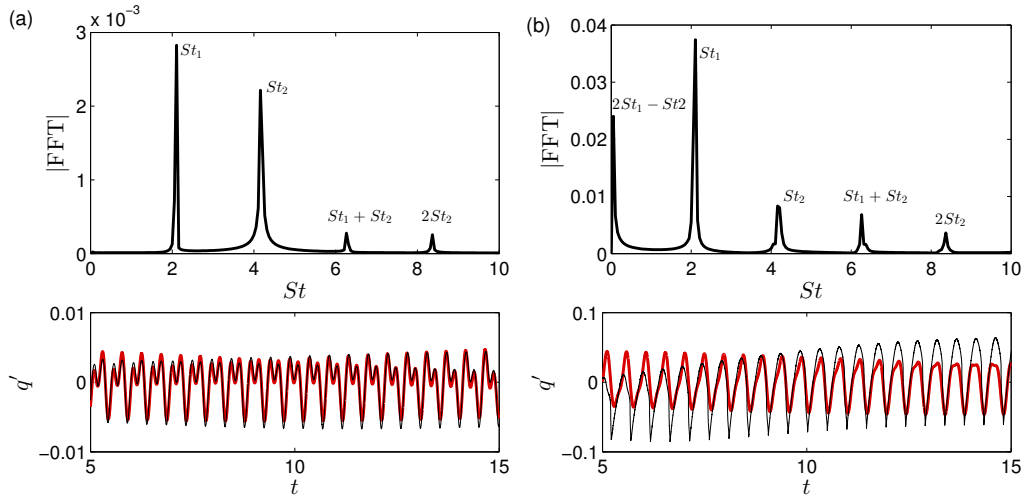


Figure 7: FFT of the heat released by the flame when the forcing is quasiperiodic with the form (11). Top frames: (a) At low forcing amplitudes,  $A_1 = 0.01$ ,  $A_2 = 0.05$ , the forcing frequencies dominate the heat release spectrum. (b) At large forcing amplitudes,  $A_1 = 0.2$ ,  $A_2 = 0.25$ , peaks at other frequencies become relevant. Bottom frames: nonlinear heat release fluctuations (thin black) and heat release reconstructed using only the peaks at the forcing frequencies (thick red).

#### 289 4. FDIDF assumptions and calculation

290 By FDIDF, we refer to the frequency domain approximation of the nonlinear flame  
291 response to a quasiperiodic velocity perturbation of the form:

$$u'_{ac} = A_1 \sin(\omega_1 t) + A_2 \sin(\omega_2 t) \quad (11)$$

292 where  $\omega_1$  and  $\omega_2$  are incommensurate frequencies. This choice guarantees that the phase  
293 between the two signals does not affect the dynamics. In the following subsections we  
294 discuss in details the approximations and assumptions we make concerning the nonlin-  
295 earity.

##### 296 4.1. FDIDF definition

297 First, as in the FDF case, we assume that the nonlinearity does not excite the sub-  
298 harmonics of the forcing frequencies, and that no intrinsic dynamical instabilities exist.  
299 Because the heat release is a nonlinear function of the forcing signal (11), we expect that  
300 its response will contain all the possible combinations of the input frequencies. By using  
301 a double Fourier series expansion [24], we can write

$$q' = \sum_m \sum_n \hat{q}_{mn} \sin[(m\omega_1 + n\omega_2)t + \phi_{mn}] \quad (12)$$

302 where the heat release amplitude coefficients  $\hat{q}_{mn}$  and the phases  $\phi_{mn}$  are functions of the  
303 input velocity frequencies and amplitudes. The integers  $m, n \in \mathbb{Z}$  are varied over all the  
304 possible combinations giving a non-negative value of the angular frequency  $m\omega_1 + n\omega_2$ .

305 In order to proceed with the harmonic balance analysis, we need to assume that  
 306 the heat release response is dominated by the frequency components at the two input  
 307 frequencies (see Fig. 4), so that it can be approximated by

$$q' \approx \hat{q}_{10} \sin(\omega_1 t + \phi_{10}) + \hat{q}_{01} \sin(\omega_2 t + \phi_{01}) \quad (13)$$

308 This assumption is less well-justified than the filtering hypothesis of the previous section,  
 309 because the latter only requires that high frequency oscillations will be damped by the  
 310 system. For the FDIDF, the coupling between the frequencies can also lead to low  
 311 frequency oscillations (e.g., at an angular frequency of  $|\omega_2 - \omega_1|$ ) for which the filtering  
 312 hypothesis does not necessarily hold. Therefore, we are implicitly assuming that the  
 313 nonlinearity's response at these frequencies is either filtered by the system or is weak.  
 314 This holds true at small forcing amplitude, for which nonlinear effects are small, but it  
 315 has to be tested at larger amplitudes.

316 Fig. 7 shows examples on the quality of the FDIDF approximations: at low forcing  
 317 amplitude (7a) nonlinear effects are weak and the heat release approximated by (13)  
 318 compares well with the fully nonlinear output. At larger input amplitudes (7b) the  
 319 quality of the approximation deteriorates. This is because the nonlinearity couples the  
 320 modes, and high peaks can be observed in the heat release FFT at frequencies which are  
 321 simple combinations of the input ones. For example, in Fig. 7b one can see that the peak  
 322 at the very low frequency  $2St_1 - St_2$  has a large amplitude, meaning that the heat release  
 323 exhibits large fluctuations over long time scales. The FDIDF approximation cannot see  
 324 these long time scale fluctuations, as shown in the bottom frame. This is because it  
 325 ignores all the FFT contributions which are not at  $St_1$  and  $St_2$ . For this reason, we  
 326 cannot expect the FDIDF method to work well at large amplitudes. Therefore, we limit  
 327 the FDIDF calculations in amplitude so that both  $A_1$  and  $A_2$  are smaller than 0.4, and  
 328 their sum is less than 0.5.

329 The FDIDF is defined as:

$$\text{FDIDF} \equiv [\mathcal{F}_{10}, \mathcal{F}_{01}] \equiv \frac{\bar{U}}{\bar{Q}} \begin{bmatrix} \hat{q}_{10} e^{i\phi_{10}} & \hat{q}_{01} e^{i\phi_{01}} \\ \hat{u}_{10} & \hat{u}_{01} \end{bmatrix} \quad (14)$$

330 where  $\hat{u}'_{10}$  and  $\hat{u}'_{01}$  are the Fourier components of the input velocity at  $\omega_1$  and  $\omega_2$  re-  
 331 spectively.  $\mathcal{F}_{10}$  ( $\mathcal{F}_{01}$ ) contains information on how the amplitude and phase of heat  
 332 release fluctuations at  $\omega_1$  ( $\omega_2$ ) vary when the flame is forced quasiperiodically. The total  
 333 (non-dimensional) heat release fluctuations are then approximated by

$$\hat{q}' \approx \text{FDIDF} \cdot [\hat{u}_{10}, \hat{u}_{01}]^T = \mathcal{F}_{10} \hat{u}_{10} + \mathcal{F}_{01} \hat{u}_{01} \quad (15)$$

334 Note that the heat release in (15) is not a simple linear superposition of two FDFs. This  
 335 is because the FDIDF's gains and phases are functions of all the four input variables  
 336 ( $A_1, \omega_1, A_2, \omega_2$ ). Finally, notice that the FDIDF is a symmetric object with respect to  
 337 the input pairs  $(A_1, \omega_1)$  and  $(A_2, \omega_2)$  so that

$$\mathcal{F}_{10}(A_1, \omega_1, A_2, \omega_2) = \mathcal{F}_{01}(A_2, \omega_2, A_1, \omega_1) \quad (16)$$

#### 338 4.2. FDIDF amplitude saturation

339 In §3, using knowledge of the flame's gain response from the literature, we performed  
 340 calculations only for frequencies close to the first two acoustic modes. No information is

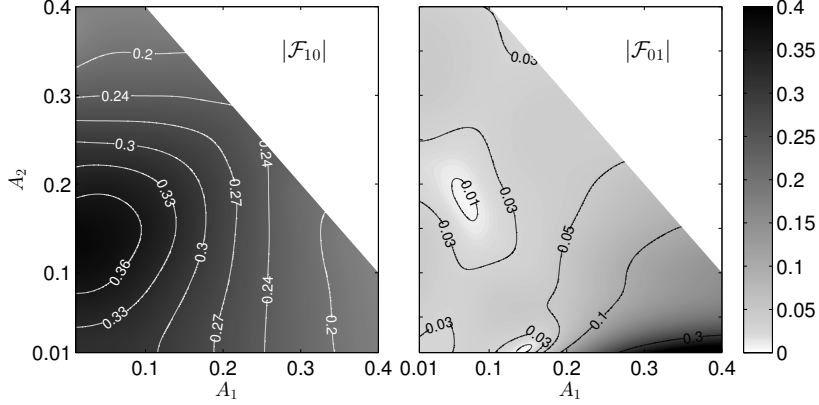


Figure 8: Amplitudes dependence of the FDIDF gains. The forcing frequencies have been fixed at the arbitrary values  $St_1 = 1.513$  and  $St_2 = 5.153$ . The region  $A_1 + A_2 > 0.5$  has not been investigated.

341 available about the gain response of conical flames when they are forced with quasiperiodic signals. However, it is reasonable to assume that, when fixing the amplitude  $A_1$   
 342 and increasing the amplitude  $A_2$  (or viceversa), the gains of the FDIDF will decrease.  
 343 This is because we expect the flame nonlinear responses  $\mathcal{F}_{10}$  and  $\mathcal{F}_{01}$  to saturate, at least  
 344 on average, with respect to the amplitudes  $A_1$  and  $A_2$  of both forcing modes. This is  
 345 proven to be correct for a simple cubic nonlinearity in [21], where also some experimental  
 346 evidence of this fact is provided.  
 347

348 We therefore assume that the FDIDF gains are less than 1. As a consequence, we  
 349 expect that self-excited thermoacoustic oscillations can only be found at frequencies for  
 350 which the acoustic gain is larger than one (see the FDIDF dispersion relations (18)).  
 351 These frequency ranges are the same as in the FDF case, because we have not modified  
 352 the acoustic system. Because we expect two modes to be unstable, it is reasonable  
 353 to guess that one of the mode's frequencies will be close to the fundamental acoustic  
 354 frequency, and the other one will be close to the acoustic second acoustic eigenfrequency.  
 355 Note that, if two modes with similar frequencies were to oscillate simultaneously, beating  
 356 phenomena could occur, and one should also investigate the coupling between these close  
 357 frequencies. However, this does not happen for the system we are considering, as was  
 358 also shown via the time domain analysis carried out in [14] on the same thermoacoustic  
 359 system. Given this, and the symmetry condition (16), we will limit the calculations  
 360 to the cases in which the non-dimensional frequencies  $St_n \equiv L_f f_n / \bar{U}$  lie in the ranges  
 361  $St_1 \in [1.273, 3.024]$  and  $St_2 \in [4.138, 7.162]$  respectively. In the following, we will  
 362 refer to mode 1 and mode 2 when referring to oscillations with a frequency in the range  
 363 spanned by  $St_1$  and  $St_2$  respectively.

#### 364 4.3. FDIDF calculation and validation

365 Fig. 8 shows an example of the FDIDF gains as a function of the two forcing ampli-  
 366 tudes. The forcing frequencies are fixed at arbitrary values. We observe that the gain of  
 367  $\mathcal{F}_{10}$  (low frequencies) is generally larger than the gain of  $\mathcal{F}_{01}$  (high frequencies); this is  
 368 in line with the low-pass filter characteristics of the conical flame we are investigating.

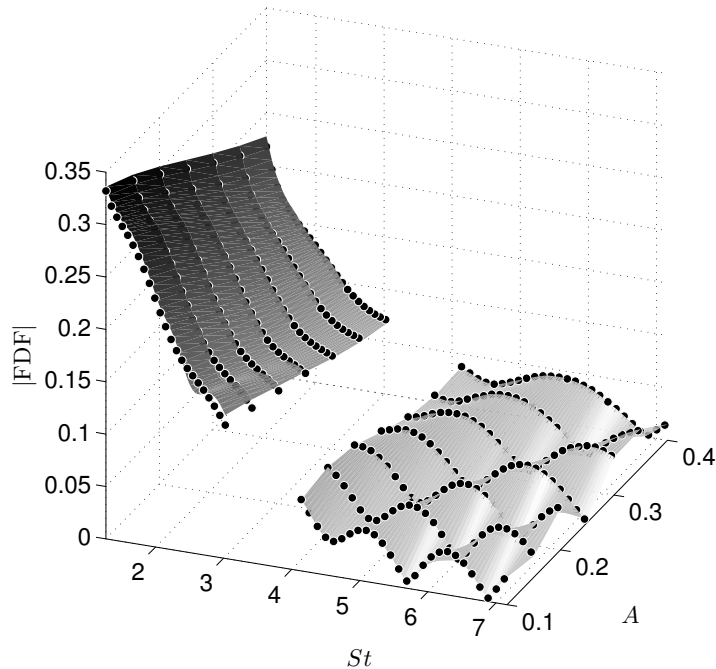


Figure 9: In the limit in which one of the two amplitude vanishes, the FDIDF tends to the FDF. FDIDF limits are plotted as surfaces and the FDF results as dots. In the region  $St \in [1.273, 3.024]$  the limit of  $|\mathcal{F}_{10}|$  is plotted fixing  $St_2 = 5.153$  and  $A_2 = 0.01$  in (17); in the region  $St \in [4.138, 7.162]$  the limit of  $|\mathcal{F}_{01}|$  is plotted fixing  $St_1 = 1.513$  and  $A_1 = 0.01$ . The results compare well over the entire set of parameters investigated.

369 Also, for  $\mathcal{F}_{10}$  we see that the gain tends to decrease with respect to both amplitudes,  
 370 as was discussed in the previous section. This is not always true for the  $\mathcal{F}_{01}$ . It is not  
 371 surprising because  $\mathcal{F}_{01}$  contains the heat release response at frequencies spanned by  $St_2$ .  
 372 Even in the FDF analysis we observed that, in this frequency range, the gain does not  
 373 decrease monotonically with the amplitude, meaning that subcritical Hopf bifurcations  
 374 and regions with multi-stable solutions may be observed.

375 In rare cases, we observe that the gain of  $\mathcal{F}_{01}$  is larger than one. This always happens  
 376 when the amplitude of  $A_1$  is large (between 0.3 and 0.4), and the amplitude of  $A_2$  is at  
 377 its minimum, 0.01. This is due to the fact that, although we numerically ensure that the  
 378 two forcing frequencies are incommensurate, their ratio can be close to a simple fraction.  
 379 For example, in some cases the frequency  $St_2$  is close to a harmonic of  $St_1$ . If the velocity  
 380 amplitude at  $St_1$  is large, the heat release responds significantly also at its harmonics.  
 381 Because we perform FFTs on signals of finite length, the FDIDF component at  $St_2$  will  
 382 see part of the harmonic contribution of  $St_1$ , artificially increasing the gain of the second  
 383 mode (see Fig. 8). This is a source of error which increases when the thermoacoustic  
 384 eigenfrequencies are close to multiples of each other. It could be reduced by integrating  
 385 the governing equations over a longer time period, in order to have a better frequency  
 386 resolution in Fourier space and distinguish the various peak contributions. However, this  
 387 would lead to an extra numerical cost, which is undesirable.

A good test to assess the accuracy of the FDIDF calculations is to look at the limit in which the amplitude of one of the two modes goes to zero. From the definitions of the FDIDF and FDF one can verify that:

$$\begin{aligned} \lim_{A_2 \rightarrow 0} \mathcal{F}_{10}(A_1, \omega_1, A_2, \omega_2) &= \text{FDF}(A_1, \omega_1) & \forall \omega_2 \\ \lim_{A_1 \rightarrow 0} \mathcal{F}_{01}(A_1, \omega_1, A_2, \omega_2) &= \text{FDF}(A_2, \omega_2) & \forall \omega_1 \end{aligned} \quad (17)$$

388 meaning that  $\mathcal{F}_{10}$  tends to the FDF results in the region covered by  $St_1$  when  $A_2$  vanishes  
 389 and, by exploiting the symmetry condition (16),  $\mathcal{F}_{01}$  tends to the FDF results in the  
 390 region covered by  $St_2$  when  $A_1$  vanishes.

391 Assuming that the FDIDF is a continuous function, we use the calculations at the  
 392 smallest amplitudes we have investigated (0.01) as limits. Therefore, the horizontal slice  
 393 of  $|\mathcal{F}_{10}|$  at  $A_2 = 0.01$  and the vertical slice of  $|\mathcal{F}_{01}|$  at  $A_1 = 0.01$  in Fig. 8 need to  
 394 match the FDF gain at  $St = 1.513$  and  $St = 5.153$  respectively (vertical slices of Fig. 5).  
 395 Fig. 9 shows this comparison over the entire range of frequencies and amplitudes we  
 396 have investigated. The limits agree well with the FDF results, with the largest difference  
 397 between the FDF and the FDIDF limit being about  $10^{-3}$ .

## 398 5. FDIDF analysis

We now couple the FDIDF with the acoustic response in a similar fashion as in Fig. 1 and find the dispersion relations that need to be satisfied for quasiperiodic oscillations to exist. The coupling between the acoustic network and the FDIDF is sketched in Fig. 10. Note that, although  $\hat{q}_{10}$  is explicitly proportional only to  $\hat{u}_{10}$  through  $\mathcal{F}_{10}$ , the latter is an implicit nonlinear function of both  $\hat{u}_{10}$  and  $\hat{u}_{01}$ . Therefore, the dispersion relations we obtain when imposing the harmonic balance condition are coupled, and need to be simultaneously satisfied:

$$\begin{aligned} \mathcal{F}_{10}(A_1, s_1, A_2, s_2)H_{x_f}(s_1) - 1 &= 0 \\ \mathcal{F}_{01}(A_1, s_1, A_2, s_2)H_{x_f}(s_2) - 1 &= 0 \end{aligned} \quad (18)$$

399 Quasiperiodic oscillations of the form (11) exist when the growth rates of the Laplace  
 400 variables  $s_n = \sigma_n + i\omega_n$  are both equal to zero, which is the condition under which the  
 401 FDIDF was calculated. However, to investigate the stability of the FDIDF solutions, we  
 402 want to calculate the rate of change of the growth rates when the calculated amplitudes  
 403 are perturbed. This will yield solutions of (18) with non-zero growth rates. Because  
 404 we are working with a non-static nonlinearity, the FDIDF is a function of two complex  
 405 variables, and it is not straightforward to extend it to the complex  $\mathbb{C}^2$  space. Thus, we  
 406 decide to use the extrusion method of [56], by assuming that  $\text{FDIDF}(A_1, s_1, A_2, s_2) =$   
 407  $\text{FDIDF}(A_1, i\omega_1, A_2, i\omega_2)$ , which is a zero-order approximation of the FDIDF around the  
 408 solutions. This complication is not present in the study of [21], because static nonlin-  
 409 earities were used. In that case, the FDIDF is a simpler object and depends only on the  
 410 forcing amplitudes, not on the frequencies.

### 411 5.1. Linear stability of limit cycles: Neimark-Sacker bifurcations

A first set of solutions of the FDIDF are those for which the amplitude of one of the two modes is equal to zero. These are the FDF harmonic solutions. For example, if



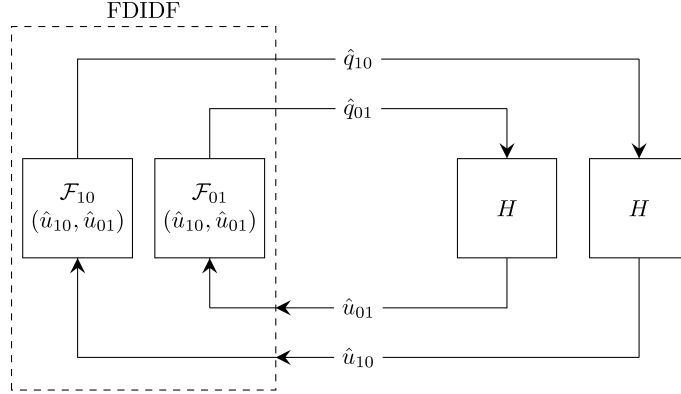


Figure 10: Sketch of the FDIDF feedback loop with the acoustics. The FDIDF (dashed block) is a two-input, two-output nonlinear object. The output is formed with the superposition of two, coupled, nonlinear elements which respond at different frequencies.

The two harmonic components of the quasiperiodic signal are indicated with subscripts  $_{10}$  and  $_{01}$  respectively. The implicit dependence of  $\mathcal{F}_{10}$ ,  $\mathcal{F}_{01}$  with respect to both  $\hat{u}_{10}$  and  $\hat{u}_{01}$  has been highlighted to emphasise that the dispersion relations (18) are coupled.

$A_2 = 0$  then we look for periodic solutions (with zero growth rate) of mode 1. From (18) we have:

$$\begin{aligned} \text{FDF}(A_1, i\omega_1)H_{x_f}(i\omega_1) - 1 &= 0 \\ \mathcal{F}_{01}(A_1, i\omega_1, 0, s_2)H_{x_f}(s_2) - 1 &= 0 \end{aligned} \quad (19)$$

412 The first equation derives from the limit (17) and converges to the FDF dispersion  
 413 relation (4). It is now decoupled from the second equation. We have already calculated  
 414 its solutions, shown as yellow lines in Fig. 6. The second dispersion relation, however,  
 415 contains information that the FDF cannot provide. It has to be solved for the frequency  
 416  $\omega_2$  and the growth rate  $\sigma_2$  by fixing the frequency and amplitude of the other mode at  
 417 the FDF solution. If the growth rate  $\sigma_2$  is positive, then oscillations at frequency  $\omega_2$  are  
 418 linearly unstable around the limit cycle with amplitude  $A_1$  and frequency  $\omega_1$ . The onset  
 419 of these instabilities is known as a secondary Hopf or Neimark-Sacker bifurcation.

420 Fig. 11 shows the bifurcation diagram of periodic solution when their stability is  
 421 assessed with the FDIDF method. Most of the limit cycles that were found to be stable  
 422 with the FDF method are now predicted to be unstable because, according to the solution  
 423 of (19), oscillations at a different frequency will grow around them. This is consistent  
 424 with the time integration results, in which we rarely observe periodic oscillations. Time  
 425 domain and FDIDF results cannot compare perfectly throughout the entire bifurcation  
 426 map, because the latter neglects contributions away from the input frequencies, which  
 427 may be important at large amplitudes. However, the FDIDF correctly captures some  
 428 of the system's bifurcations. For example, analysing Fig. 11 from  $x_f = 0.60$  backwards,  
 429 time marching results show a supercritical Hopf bifurcation at  $x_f = 0.59$ , a Neimark-  
 430 Sacker bifurcation at  $x_f = 0.50$ , and an inverse Neimark-Sacker at  $x_f = 0.11$ . The

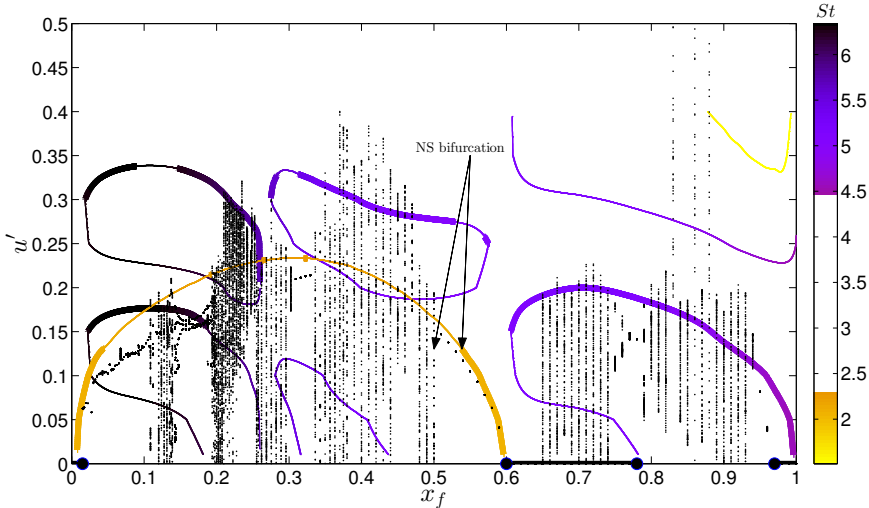


Figure 11: Comparison between time domain (as in Fig. 6) and FDIDF period bifurcation diagrams. The flame position  $x_f$  is used as a control parameter. Thin and thick lines are used to plot unstable and stable limit cycles respectively. Neimark-Sacker bifurcations are found at the edges of the stable solution with a non-zero amplitude.

431 FDIDF method locates correctly the first Hopf bifurcation for mode 1, and predicts  
 432 Neimark-Sacker bifurcations at  $x_f = 0.53$  and  $x_f = 0.045$ .

433 With the FDIDF we can also calculate the frequency of oscillations that grow around  
 434 limit cycles after Neimark-Sacker bifurcations. At  $x_f = 0.53$  the FDIDF predicts that oscil-  
 435 lations with a non-dimensional frequency  $St_2 = 5.0136$  are linearly unstable ( $\sigma_2 = 2.86 \cdot 10^{-5}$ )  
 436 around the limit cycle with  $A_1 = 0.1305$  and  $St_1 = 2.1132$ . This prediction can be com-  
 437 pared with self-excited time domain results. Fig. 12 shows the FFT of the velocity signal  
 438 just before and after the Neimark-Sacker bifurcation in the time domain (see Fig. 11). In  
 439 the former case, the oscillation is dominated by a component at frequency  $St_1 = 2.148$   
 440 with intensity  $A_1 = 0.1399$ . Just after the bifurcation a second high peak appears at  
 441  $St_2 = 5.005$ . All these results are consistent with the FDIDF predictions.

#### 442 5.1.1. Discussion on cost and practical implementation

443 The FDIDF is a function of four independent input parameters. As a consequence,  
 444 the numerical cost of building such an object increases quickly when wide ranges of  
 445 parameters are investigated. By using the arguments in §§4.1-4.2, we limit as much  
 446 as possible the width of these ranges. However, it is non-trivial to determine how to  
 447 discretize these regions to appropriately estimate the FDIDF response. Because our  
 448 model is low-order, we can afford to carry out a very detailed calculation of the FDIDF,  
 449 and then investigate its dependence on the number of points used. We use about 60  
 450 discretization points for each frequency range. We vary the amplitudes in the range  
 451  $[0.01, 0.4]$  in 9 steps, with the additional constraint that their sum does not exceed the  
 452 threshold value of 0.5. With these limits, the total number of simulations we run to build  
 453 the FDIDF is about 200 000. About 70 000 CPU hours were required to perform the  
 454 analysis, which is approximately 5 times more expensive than the continuation method

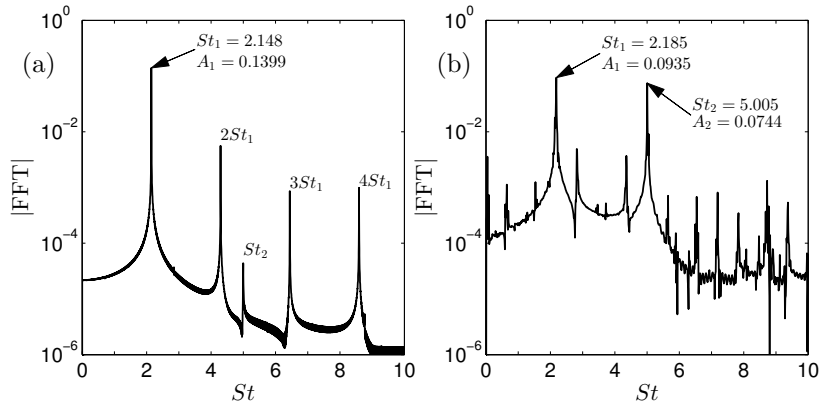


Figure 12: FFT of velocity fluctuations of time domain simulations as described in [14]. (a):  $x_f = 0.51$ , the solution is periodic, only a dominant peak at  $St_1 = 2.148$  is found; its harmonics are present but negligible. (b)  $x_f = 0.49$ , a second intense peak is found at  $St_2 = 5.005$ ; the system has undergone a subcritical Neimark-Sacker bifurcation.

455 used by [12] to calculate limit cycle bifurcations on a similar thermoacoustic system  
 456 with a continuation algorithm. We then use a four-dimensional cubic spline method to  
 457 interpolate the real and imaginary parts of the FDIDF between calculated points onto a  
 458 much finer grid, as functions of the input amplitudes and frequencies. The interpolation  
 459 is performed both using the full set of simulations or partial information only, to assess  
 460 the effect of the discretization on the system dynamics.

461 Fig. 13 shows the relative error of the interpolated FDIDF as a function of the number  
 462 of points (always uniformly spaced) used for the interpolation. The results at the finest  
 463 discretization, which are those used for the analysis in the rest of this study, are used as  
 464 reference. By halving the number of discretization points used for  $St_1$  and  $St_2$ , the cost  
 465 of the FDIDF is reduced by a factor of 4 and the percentage error is about 5%. However,  
 466 further reduction in the number of points used for the interpolation lead to larger errors,  
 467 and significant deviation from the actual dynamic response should be expected. This  
 468 shows that a large number of calculations is required to accurately estimate the FDIDF.  
 469 This makes it currently non-affordable for, say, compressible LES studies, in which many  
 470 CPU hours are already required to calculate the FDF only [57].

471 Part of the high cost of the current FDIDF analysis is due to the fact that all possible  
 472 flame positions are investigated. Because a temperature jump follows the flame, the  
 473 eigenfrequencies vary significantly when  $x_f$  spans from 0 to 1, and wide range of frequencies  
 474 need to be investigated. In practical situations this is probably not the case, and the  
 475 frequency bands of interest may be narrower, thus reducing the number of calculations  
 476 required for the FDIDF. Also, we emphasise that, to calculate the stability of limit cycles  
 477 found with the FDF (as was shown in §5.1), we need only a part of the FDIDF calcula-  
 478 tion. This is because we examine cases in which one of the two amplitudes is small.  
 479 The only parameter that has to be varied is the frequency of the small amplitude mode.  
 480 In this framework, the FDIDF method is much cheaper (it approximately reduces to the  
 481 cost of two FDFs), and is comparable in cost with the continuation method described

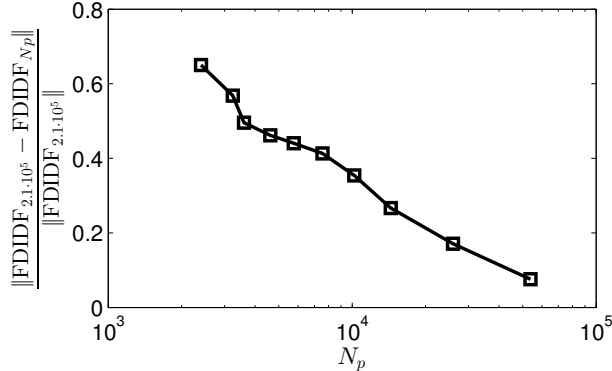


Figure 13: Interpolated FDIDF relative error dependence with respect to the number of discretization points  $N_p$  used. Choosing fewer than 30 000 points leads to deviations from the actual response larger than 10%.

482 by [12]. The latter remains more accurate, because it studies the stability of periodic  
 483 solutions (i.e., the spectrum of the oscillations may contain peaks at the harmonics of  
 484 the fundamental frequency), whereas the FDIDF is limited to harmonic solutions (i.e.,  
 485 the spectrum of the oscillations contains only one peak at the fundamental frequency).  
 486 The advantage of the FDIDF is that it can be reused in different acoustic networks to  
 487 calculate the stability of several thermoacoustic systems.

488 The use of the FDIDF to assess the stability of periodic solutions could also be  
 489 exploited in experiments at approximately the cost of two FDFs by means of the following  
 490 procedure: (i) measure an FDF; (ii) obtain harmonic solutions and their stability – with  
 491 respect to a single mode – with the harmonic balance; (iii) for solutions that are predicted  
 492 to be stable by the FDF method, perform another set of experiments to assess again their  
 493 stability with respect to other forcing frequencies. This is accomplished by forcing the  
 494 flame with a signal of the form (11), by fixing the amplitude and frequency of a mode  
 495 at the FDF solution and the amplitude of the other mode at a small value. The only  
 496 parameter left is the frequency of the second mode. It has to be varied over a range  
 497 of dangerous frequencies, which can be obtained by the FDF results and the acoustic  
 498 response. The stability of the FDF solutions with respect to other frequencies can then  
 499 be calculated following the procedure described in §5.1.

## 500 5.2. Prediction and stability of quasiperiodic oscillations

501 Once limit cycles have become unstable, thermoacoustic oscillations converge towards  
 502 another stable solution. This can be another periodic solution, with a different frequency  
 503 and amplitude, a quasiperiodic attractor, or even a strange attractor. The FDIDF can  
 504 approximate the location and stability of periodic and quasiperiodic solutions, but cannot  
 505 predict the existence of other types of attractors, which were shown to exist in this type  
 506 of thermoacoustic system by [11, 14].

507 When looking for quasiperiodic attractors, the dispersion relations (18) need to be  
 508 solved by fixing the growth rates  $\sigma_1$  and  $\sigma_2$  at zero, and looking for solutions with finite  
 509 amplitudes for both modes. We rely on numerical techniques to find the roots of (18)  
 510 that satisfy this conditions starting from a good initial guess. Because five parameters

511 (two amplitudes, two frequencies and the bifurcation parameter) can be varied, a large  
512 number of initial guesses can be chosen, which is numerically inefficient. To reduce the  
513 numerical cost, we first locate isolated solutions by starting from a coarse grid of initial  
514 guesses that covers the parameter space. Then, we extend the solutions to continuous  
515 lines with an arclength continuation method by slowly varying the bifurcation parameter.

### 516 5.2.1. Stability criterion

517 We find several sets of quasiperiodic solutions and require a criterion to assess their  
518 stability. From a dynamical system viewpoint, the coupled evolution of the oscillations'  
519 amplitudes can be written in terms of a linear operator  $L$  and a nonlinear operator  $N(\mathbf{A})$   
520 as

$$\frac{dA_j}{dt} = L_j A_j + N_j(\mathbf{A}) \equiv \sigma_j(\mathbf{A}) A_j \quad (20)$$

where  $\mathbf{A}$  is the amplitudes vector. Although the explicit expressions for the linear and nonlinear operators are not known,  $\sigma_j$  represents a nonlinear growth rate, in the sense that its intensity varies with the amplitudes of the oscillations. When at least one  $\sigma_j$  equals zero, a non-trivial solution (with a finite amplitude) to the dynamical system has been found. The amplitude of each mode varies with respect to the value of its growth rate only, which is implicitly a function of all the amplitudes. For our system, which contains only two modes, the dynamical system (20) reduces to:

$$\begin{aligned} \dot{A}_1 &= \sigma_1(A_1, A_2) A_1 \\ \dot{A}_2 &= \sigma_2(A_1, A_2) A_2 \end{aligned} \quad (21)$$

521 Equations (21) will slowly vary the oscillations' amplitudes, which in turn will change the  
522 growth rates and frequencies according to the solution of (18) at the current amplitudes.  
523 Eqs. (21) were also discussed in [21], where their interpretation in terms of an averaging  
524 procedure was also provided.

525 Let us now indicate a solution of (21) with overlines. These solutions are fixed  
526 point if both amplitudes are equal to zero, limit cycles if only one amplitude is zero, or  
527 quasiperiodic if both amplitudes are non-zero. By linearising eq.(21) around a solution  
528 the evolution of small perturbations, indicated with  $\Delta$ , is given by:

$$\frac{d}{dt} \begin{bmatrix} \Delta A_1 \\ \Delta A_2 \end{bmatrix} = \begin{bmatrix} \frac{\partial \sigma_1}{\partial A_1} \bar{A}_1 + \bar{\sigma}_1 & \frac{\partial \sigma_1}{\partial A_2} \bar{A}_1 \\ \frac{\partial \sigma_2}{\partial A_1} \bar{A}_2 & \frac{\partial \sigma_2}{\partial A_2} \bar{A}_2 + \bar{\sigma}_2 \end{bmatrix} \begin{bmatrix} \Delta A_1 \\ \Delta A_2 \end{bmatrix} \equiv \mathbf{J} \Delta \mathbf{A} \quad (22)$$

529 where the partial derivatives are evaluated at the solution. If the eigenvalues of the  
530 Jacobian  $\mathbf{J}$  have negative real parts, the solution under consideration is stable.

531 It is worth discussing the forms that the Jacobian assumes for the different types of  
532 solutions. For a fixed point, both amplitudes vanish and  $\mathbf{J}$  simply contains the growth  
533 rates  $\sigma_1$  and  $\sigma_2$  on the main diagonal, retrieving the classic linear stability result. For a  
534 limit cycle solution (say of mode 1), the Jacobian takes the form

$$\mathbf{J}_{LC} = \begin{bmatrix} \frac{\partial \sigma_1}{\partial A_1} \bar{A}_1 & \frac{\partial \sigma_1}{\partial A_2} \bar{A}_1 \\ 0 & \bar{\sigma}_2 \end{bmatrix} \quad (23)$$

535 and has eigenvalues  $\partial \sigma_1 / \partial A_1 \bar{A}_1$  and  $\bar{\sigma}_2$ . Because  $\bar{A}_1$  is positive, the stability is deter-  
536 mined by the sign of  $\partial \sigma_1 / \partial A_1$  (the FDF condition) and  $\sigma_2$ . This corresponds to the

537 stability condition that was intuitively discussed in the previous section. Furthermore,  
 538 the eigenvector corresponding to the limit cycle eigenvalue  $\partial\sigma_1/\partial A_1 \bar{A}_1$  is orientated along  
 539 the  $A_1$  direction. The second eigenvector, however, has a non-trivial direction and can  
 540 be calculated only having the FDIDF. We will shortly return to the significance of these  
 541 eigenvectors in the FDIDF analysis.

542 Lastly, for quasiperiodic solutions we obtain that the stability is determined by the  
 543 eigenvalues of the Jacobian

$$\mathbf{J}_{QP} = \begin{bmatrix} \frac{\partial\sigma_1}{\partial A_1} \bar{A}_1 & \frac{\partial\sigma_1}{\partial A_2} \bar{A}_1 \\ \frac{\partial\sigma_2}{\partial A_1} \bar{A}_2 & \frac{\partial\sigma_2}{\partial A_2} \bar{A}_2 \end{bmatrix} \quad (24)$$

544 This is not exactly the condition that was suggested by [21], whose Jacobian does not  
 545 depend on the solution amplitudes. Nonetheless, condition (24) derives from the linearisation  
 546 of the amplitudes' evolution around a solution. Given that we retrieve correct  
 547 physical conditions for the stability of fixed point and limit cycles, we shall expect it to  
 548 hold even for quasiperiodic oscillations.

549 Two methods can be used to calculate the partial derivatives of the growth rates  
 550 with respect to the amplitudes. By brute force, in analogy with eq. (10), one can fix one  
 551 amplitude at its solution's value, slightly perturb the other amplitude, and determine the  
 552 variations in frequency and growth rate of both modes by solving (18) with an iterative  
 553 method. Alternatively, the implicit function theorem may be used, as suggested by [21].  
 554 The latter is quicker and more reliable because no iterative methods need to be used.  
 555 Details on the implicit function theorem method are given in Appendix A. Both methods  
 556 have been tested and yield the same results.

### 557 5.2.2. FDIDF bifurcation analysis

558 Fig. 14 contains the FDIDF solutions when the bifurcation parameter is varied between  
 559  $0.40 \leq x_f \leq 0.60$ . Limit cycle solutions lie on  $A_1 = 0$  and  $A_2 = 0$ , whereas  
 560 solutions for which both amplitudes are non-zero are quasiperiodic. We have plotted  
 561 with black filled circles attractors (solutions for which both eigenvalues have a negative  
 562 real part), with red empty circles repellors (both eigenvalues have a positive real part),  
 563 and with red squares saddle-nodes (one eigenvalue has a positive real part, and the other  
 564 a negative real part). The latter are particularly interesting because thermoacoustic oscillations  
 565 can be first attracted towards them along their stable manifold, and only later  
 566 diverge along the unstable manifold towards an attractor. If the growth rate of the unstable  
 567 mode is small, the oscillations may persist for a long time around the saddle-node  
 568 state. This can be problematic for time domain simulations or experiments, because the  
 569 system has to be observed for a long time before being sure that the final attractor has  
 570 been reached. Saddle-nodes in thermoacoustic systems were also discussed in [11], where  
 571 they were referred to as "unstable attractors".

572 A convenient way of representing the FDIDF results is through phase-planes. A  
 573 phase-plane contains the trajectories that the amplitudes will follow before converging  
 574 to an attractor. Starting from different initial conditions can lead thermoacoustic oscillations  
 575 towards different attractors. The set of initial conditions that converge towards  
 576 an attractor is known as the basin of attraction of the attractor. On a theoretical basis,  
 577 it should be possible to identify the basins of attraction boundaries by investigating  
 578 the growth rates of the thermoacoustic modes while varying the oscillations' amplitudes.

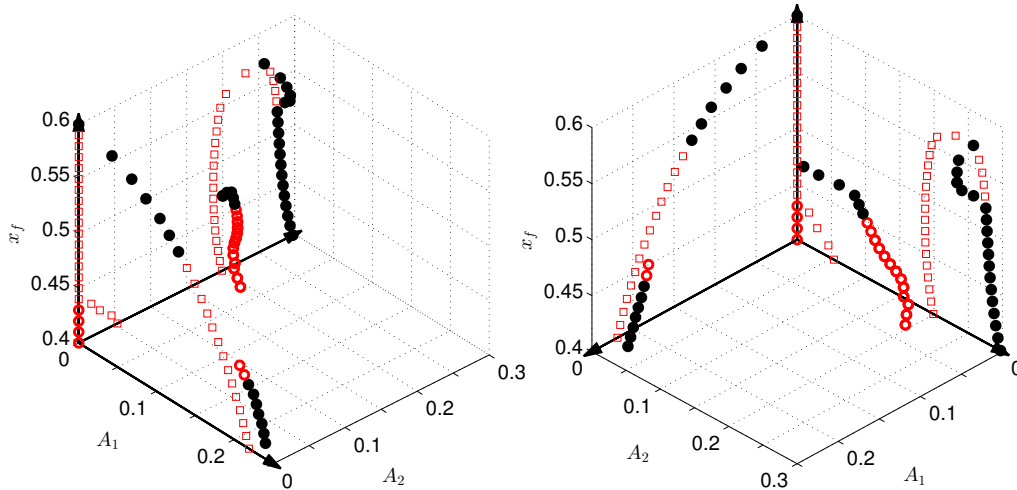


Figure 14: FDIDF bifurcation diagram in the region  $0.40 \leq x_f \leq 0.60$ . The FDF solutions of mode 1 and 2 lie on the  $A_2 = 0$  and  $A_1 = 0$  planes, respectively. The fixed point solutions lie on the line  $A_1 = A_2 = 0$ . Super- and subcritical quasiperiodic oscillations are found. The stability of all solutions is assessed with the FDIDF conditions. Stable attractors are indicated with filled black circles, repellers with empty red circles and saddle-nodes with empty red squares. Two views of the same bifurcation diagram are shown.

579 However we find that when we are not close to solutions of our system, the growth rates  
580 quickly become large. The FDIDF was not evaluated under these conditions, therefore  
581 it cannot be used to build the phase-planes because the extrusion method we adopted is  
582 no longer valid. Note that, for a static nonlinearity as the one considered by [21], this  
583 problem does not arise because the FDIDF is a function of the amplitudes only.

584 For the non-static nonlinearity we are considering in this study, the FDIDF can still be  
585 used to estimate the phase-planes. This is accomplished by calculating the eigenvectors  
586 of the Jacobian (22). By means of the Centre Manifold Theorem [58], the eigenspaces  
587 spanned by the eigenvectors associated with the stable and unstable eigenvalues are  
588 locally tangent to the stable and unstable manifolds respectively. A sketch of the phase-  
589 planes of our system across the Neimark-Sacker bifurcation at  $x_f = 0.53$  is shown in  
590 Fig. 15. Stable and unstable solutions are plotted with the same shape and colour  
591 scheme of Fig. 14, together with vectors pointing in the direction of their eigenvectors.  
592 For saddle-nodes, these vectors are locally tangent to the stable and unstable manifolds.  
593 For attractors and repellers, the eigenvalues and eigenvectors of  $\mathbf{J}$  can be complex-valued.  
594 In this case, trajectories will spiral inwards/outwards the solution. We have also sketched  
595 with dashed lines possible heteroclinic orbits. A heteroclinic orbit is a path that connects  
596 an unstable solution to a stable one. Note that some solutions may be missing from our  
597 maps, because they can lie in a region we have not investigated (large amplitudes or  
598 amplitudes smaller than 0.01), or they can be strange attractors that we cannot locate.

599 Although we have only partial information about phase-planes, they help to identify  
600 possible routes that thermoacoustic oscillations undertake before converging to an  
601 attractor. For example, let us consider Fig. 15a, which corresponds to the  $x_f$  location  
602 just before the Neimark-Sacker bifurcation marked in Fig. 11. Starting from the quies-

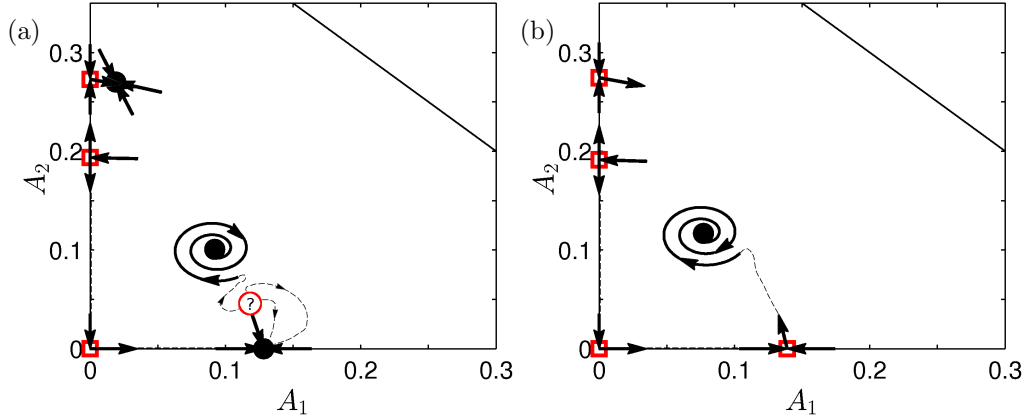


Figure 15: Sketch of phase-planes before (a) and after (b) the Neimark-Sacker bifurcation at  $x_f = 0.53$ . The arrows pointing inwards and outwards the solutions indicate the directions of the Jacobian eigenvectors with a negative and positive growth rate, respectively. Some hypothetical heteroclinic orbits are sketched with dashed lines. Across the bifurcation, the amplitude  $A_1$  ( $A_2$ ) of the attractor reached starting from a quiescent state suddenly decreases (increases).

603 cent state  $A_1 = A_2 = 0$ , the oscillations will be attracted towards the stable limit cycle  
 604 solution along  $A_2 = 0$ . However, starting from an excited state with  $A_2 \neq 0$  or by trig-  
 605 gering the system, the oscillations may converge to a different attractor. Here, the other  
 606 possible attractors are quasiperiodic. We have added a non-calculated solution (marked  
 607 with a question mark) in order to sketch some heteroclinic paths. The position of this  
 608 solution however is not entirely arbitrary; by slowly varying the bifurcation parameter,  
 609 the limit cycle solution at  $A_2 = 0$  loses its stability. A possible scenario is that, at the  
 610 bifurcation point, a quasiperiodic repeller (or another type of oscillation) collapses onto  
 611 the stable limit cycle solution. After the bifurcation (Fig. 15b), the limit cycle on  $A_1$   
 612 turns into a saddle-node, changing the topology of the phase-plane. Now, starting from  
 613 a quiescent state thermoacoustic oscillations are first attracted towards the limit cycle  
 614 solution along the  $A_2 = 0$  axis (which is the limit cycle stable manifold), and only later  
 615 are repelled from it along the unstable manifold towards the stable quasiperiodic oscilla-  
 616 tion. This is exactly what is observed in time domain simulations, although not shown  
 617 here. Analogous time domain results can be find in [11].

618 Lastly, Fig. 15 also shows that the amplitudes  $A_1$  and  $A_2$  suddenly vary across the  
 619 bifurcation. This is possible across a Neimark-Sacker bifurcation, as solutions are sud-  
 620 denly attracted towards a different attractor. Time domain results of the same bifurca-  
 621 tion shown in Fig. 12 are in line with this FDIDF prediction. Indeed, The FFT of the  
 622 time signal before and after the bifurcation shows that the amplitudes  $A_1$  ( $A_2$ ) suddenly  
 623 decreases (increases) across the bifurcation. This feature of Neimark-Sacker bifurcations  
 624 is also seen in the time domain results shown in Fig. 11. At  $x_f = 0.50$  the maximum  
 625 amplitude of the oscillations suddenly deviates from the limit cycle amplitude before  
 626 the bifurcation. A fair comparison between the oscillation amplitudes predicted by the  
 627 FDIDF and time marching is seen by looking at the position of the stable quasiperiodic



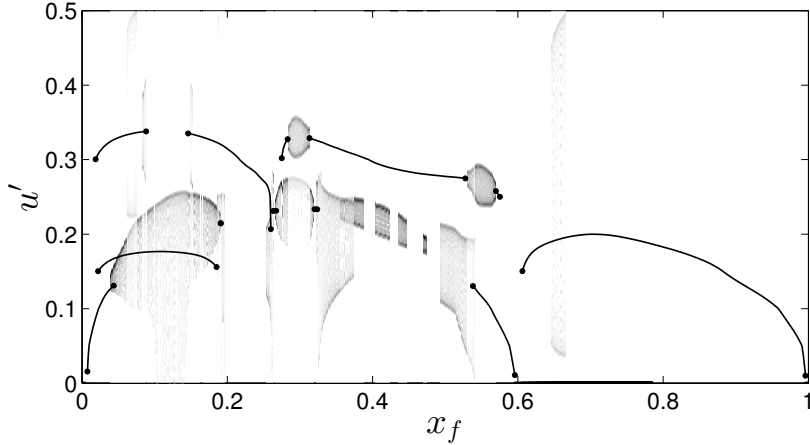


Figure 16: Overview of the FDIDF bifurcation diagram. PDF of stable quasiperiodic solutions' peaks (shaded regions) are plotted on top of stable limit cycle amplitudes (lines). The PDF intensity is higher in darker regions. The location of Neimark-Sacker bifurcations is highlighted with dots.

628 attractor in Fig. 15 and the intensity of the peaks in Fig. 12. Note that, however, these  
 629 figures contain information at slightly different values of  $x_f$ , because the location of the  
 630 Neimark-Sacker bifurcation predicted by the two methods is slightly different, due to the  
 631 FDIDF approximations.

632 We conclude this study by showing in Fig. 16 the bifurcation diagram calculated  
 633 with the FDIDF in the entire range  $0 \leq x_f \leq 1$ . We have plotted only the peaks of  
 634 stable oscillations, which are those observable in self-excited experiments or time domain  
 635 simulations. For limit cycle solutions, these peaks are shown as lines corresponding  
 636 at the oscillations amplitude. For quasiperiodic solutions, we calculate the Probability  
 637 Density Functions (PDF) of their peaks. These are shown as shaded regions in Fig. 16.  
 638 The locations of Neimark-Sacker bifurcations have been highlighted with black dots.  
 639 One can see that there is a nice match between their location and the onset of stable  
 640 quasiperiodic solutions. This does not happen if we apply the stability criterion for  
 641 quasiperiodic solutions contained in [21]. In the region  $0.60 \leq x_f \leq 1$ , quasiperiodic  
 642 oscillations tend to have a large amplitude, which exceeds the  $A = 0.5$  threshold we have  
 643 set when calculating the FDIDF. This is partly consistent with time domain results, in  
 644 which very large oscillations, e.g. at  $x_f = 0.83$ , are occasionally observed. In some  
 645 regions, multiple stable solutions are found. With time marching methods, a thorough  
 646 investigation of the initial condition is required to find these solutions.

## 647 6. Summary and conclusions

648 We have presented a numerical approach for the investigation of non-periodic ther-  
 649 moacoustic oscillations. A Flame Double Input Describing Function (FDIDF) of a non-  
 650 static nonlinear flame model based on the  $G$ -equation has been calculated by forcing  
 651 the flame with a quasiperiodic signal composed of two harmonic components with in-  
 652 dependent amplitudes and incommensurate frequencies. The FDIDF assumptions and

653 limitations have been outlined, and it has been tested against the Flame Describing Func-  
654 tion (FDF) in the limit in which the amplitude of a mode is small. The FDIDF has been  
655 embedded into a thermoacoustic network and, through the harmonic balance method,  
656 stable and unstable thermoacoustic oscillations have been calculated. Furthermore, a  
657 criterion to assess their stability has been derived.

658 The FDIDF contains a far more accurate approximation of the nonlinear flame re-  
659 sponse than the FDF. Exploiting all its information, one can predict the amplitude and  
660 stability of quasiperiodic solutions. Also, via the Centre Manifold Theorem, it can be  
661 used to sketch phase-planes to understand the path that thermoacoustic oscillations tra-  
662 jectories will take. Quantitative comparisons between the FDIDF and time marching  
663 results have been presented. We have shown that the FDIDF is capable of predicting  
664 the location of Neimark-Sacker bifurcations, the frequency of the unstable modes and  
665 the amplitude of the final quasiperiodic oscillations. We have discussed in detail the  
666 change in behaviour of a system at a Neimark-Sacker bifurcation, across which a new  
667 mode becomes unstable and the amplitude of the oscillations varies abruptly. This can  
668 lead to quasiperiodic oscillations or mode-switching to another stable periodic oscillation  
669 at a different frequency. Neither type of behaviour can be predicted by linear stability  
670 analysis nor by the FDF framework.

671 Although the FDIDF is an expensive object to calculate, for simple dynamical flame  
672 models, such as the  $G$ -equation, this is affordable. Also, we have shown how its cost can  
673 be greatly reduced if one is interested in calculating only the stability of limit cycles.  
674 This accounts for the nonlinear interaction between modes, which the FDF ignores, and  
675 provides the location of Neimark-Sacker bifurcations. Only the information at which  
676 one of the amplitudes is fixed at a very small value is needed for this, and the cost of  
677 the FDIDF reduces to the cost of a second FDF, making it affordable for experimental  
678 purposes too. We find that, for our system, most of the limit cycles that are predicted to  
679 be stable by the FDF method, are predicted to be unstable by the FDIDF method. This  
680 is consistent with self-excited time marching results of the same thermoacoustic system.  
681 Within this framework, the FDIDF is capable of predicting the frequency of oscillations  
682 that will grow in time around limit cycles. Knowing these frequencies, Helmholtz res-  
683 onators can be tuned and retro-fitted to the thermoacoustic system in order to make it  
684 less prone to oscillations.

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## 688 Appendix A. Growth rate variations by implicit function theorem

For convenience, let us rewrite the dispersion relations (18) in a compact form by  
splitting them into real and imaginary part as  $\mathbf{N}(\mathbf{A}, \mathbf{y}) \equiv (N_{10}^{\text{Re}}, N_{10}^{\text{Im}}, N_{01}^{\text{Re}}, N_{01}^{\text{Im}}) = \mathbf{0}$ ,

where we have defined

$$\begin{aligned}
N_{10}^{\text{Re}} &\equiv \text{Re} [\mathcal{F}_{10}(A_1, \omega_1, A_2, \omega_2, )H(\sigma_1 + i\omega_1) - 1] \\
N_{10}^{\text{Im}} &\equiv \text{Im} [\mathcal{F}_{10}(A_1, \omega_1, A_2, \omega_2, )H(\sigma_1 + i\omega_1) - 1] \\
N_{01}^{\text{Re}} &\equiv \text{Re} [\mathcal{F}_{01}(A_1, \omega_1, A_2, \omega_2, )H(\sigma_2 + i\omega_2) - 1] \\
N_{01}^{\text{Im}} &\equiv \text{Im} [\mathcal{F}_{01}(A_1, \omega_1, A_2, \omega_2, )H(\sigma_2 + i\omega_2) - 1]
\end{aligned} \tag{A.1}$$

where  $\mathbf{A} \equiv (A_1, A_2)$  is the vector of amplitudes, and  $\mathbf{y} \equiv (\sigma_1, \omega_1, \sigma_2, \omega_2)$  is the vector of growth rates and frequencies. This is a system of four equations through which the four dependent variables (frequencies and growth rates) are implicit functions of the amplitude levels, i.e.,  $\mathbf{y} = \mathbf{y}(\mathbf{A})$ . By implicit differentiation of the dispersion relations, one obtains

$$d\mathbf{N} = \frac{\partial \mathbf{N}}{\partial \mathbf{y}} d\mathbf{y} + \frac{\partial \mathbf{N}}{\partial \mathbf{A}} d\mathbf{A} = \mathbf{0} \tag{A.2}$$

or, by rearranging

$$\frac{d\mathbf{y}}{d\mathbf{A}} = \begin{bmatrix} \frac{\partial \sigma_1}{\partial A_1} & \frac{\partial \sigma_1}{\partial A_2} \\ \frac{\partial \omega_1}{\partial A_1} & \frac{\partial \omega_1}{\partial A_2} \\ \frac{\partial \sigma_2}{\partial A_1} & \frac{\partial \sigma_2}{\partial A_2} \\ \frac{\partial \omega_2}{\partial A_1} & \frac{\partial \omega_2}{\partial A_2} \end{bmatrix} = - \left( \frac{\partial \mathbf{N}}{\partial \mathbf{y}} \right)^{-1} \frac{\partial \mathbf{N}}{\partial \mathbf{A}} \tag{A.3}$$

The latter expression yields the growth rates and frequencies sensitivities with respect to amplitudes variations. The right hand side terms can be evaluated by finite difference by imposing small perturbations (one by one) in the dispersion relations (A.1). No iterative methods are required when using the implicit function theorem, which makes the method more reliable because is not susceptible to convergence problems.

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