Supplementary Material for

On the morphology of non-spherical particles using tandem aerodynamic diameter, mobility diameter, and mass measurements

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S1. Uncertainty analysis

The uncertainty analysis of the reported results in the present study was conducted based on ANSI/ASME Measurement Uncertainty Standard (Abernethy et al., 1985). In calculating the total uncertainty, two types of uncertainties should be considered: (i) precision uncertainty (also known as random error) and (ii) bias uncertainty (also known as systematic or instrument error). The precision uncertainty is calculated from repeated measurements and, for a small number of samples (n < 30), is calculated from the following equation

$$P_x = t_{1-c,n-1} \frac{o_n}{\sqrt{n}} \tag{S1}$$

where P_x is the precision uncertainty in quantity x, n is the number of samples (repeated measurements), $t_{1-c,n-1}$ is the Student's *t*-distribution variable at confidence interval of c (95% in this study) and degree of freedom of n-1, and σ_n is the standard deviation of samples (square-root of the variance).

The bias uncertainty, B_x , is the error of the instrument to read the correct value of a measurement. When a reported parameter depends on two or more independent variables as $f(x_1, x_2, x_3, ...)$, the propagation of uncertainty is used as follows:

$$B_{x} = \sqrt{\left(\Delta x_{1} \frac{\partial f}{\partial x_{1}}\right)^{2} + \left(\Delta x_{2} \frac{\partial f}{\partial x_{2}}\right)^{2} + \left(\Delta x_{3} \frac{\partial f}{\partial x_{3}}\right)^{2} + \cdots}$$
(S2)

where Δ denotes the uncertainty in the corresponding independent variable. The total uncertainty, U_x , is then estimated as follows

$$U_x = \sqrt{P_x^2 + B_x^2} \tag{S3}$$

The instruments used for the measurement of effective density had the following bias uncertainties: 3% in particle mobility diameter measurement using the DMA (Kinney et al., 1991), 2.7% in mass measurement of singly-charged particles using the CPMA (Symonds et al., 2013), and 4.3% in particle aerodynamic diameter measurement using the AAC (Tavakoli & Olfert, 2014).

To calculate the bias uncertainty in effective density using a combination of any two instruments, the principle of propagation of bias uncertainty was used. Using the DMA-CPMA method, the particle effective density is determined using the following equation as noted in Section 2 of the manuscript

$$\rho_{\rm eff} = \frac{6m}{\pi d_{\rm m}^3} \tag{S4}$$

and its bias uncertainty, $\Delta \rho_{\rm eff}$, is calculated as

$$\frac{\Delta\rho_{\rm eff}}{\rho_{\rm eff}} = \sqrt{\left(\frac{\Delta m}{m}\right)^2 + 9\left(\frac{\Delta d_{\rm m}}{d_{\rm m}}\right)^2} = \sqrt{(0.027)^2 + 9(0.03)^2} \sim 9.4\%$$
(S5)

where Δ denotes the uncertainty (error) in the corresponding physical quantity.

Using the AAC-DMA method, the particle effective density is determined using the following equation as noted in Section 2 of the manuscript

$$\rho_{\rm eff} = \rho_0 \frac{d_{\rm a}^2 C_{\rm c}(d_{\rm a})}{d_{\rm m}^2 C_{\rm c}(d_{\rm m})} \tag{S6}$$

The mean uncertainty in Cunningham slip correction factor, C_c , for particle diameter in the range of 20–270 nm is ~2.5% (Kim et al., 2005). Thus, the bias uncertainty in effective density is calculated as

$$\frac{\Delta\rho_{\rm eff}}{\rho_{\rm eff}} = \sqrt{4\left(\frac{\Delta d_{\rm a}}{d_{\rm a}}\right)^2 + \left(\frac{\Delta C_{\rm c}(d_{\rm a})}{C_{\rm c}(d_{\rm a})}\right)^2 + 4\left(\frac{\Delta d_{\rm m}}{d_{\rm m}}\right)^2 + \left(\frac{\Delta C_{\rm c}(d_{\rm m})}{C_{\rm c}(d_{\rm m})}\right)^2} = \sqrt{4(0.043)^2 + (0.025)^2 + 4(0.03)^2 + (0.025)^2} \sim 11.1\%.$$
(S7)

Using the AAC-CPMA method, the particle effective density is determined using the following equation as noted in Section 2 of the manuscript

$$\rho_{\rm eff} = \left(\frac{\pi d_{\rm a}^3}{6m}\right)^2 \left(\frac{\rho_0 C_{\rm c}(d_{\rm a})}{C_{\rm c}(d_{\rm m})}\right)^3 \tag{S8}$$

and the bias uncertainty in effective density is calculated as

$$\frac{\Delta\rho_{\rm eff}}{\rho_{\rm eff}} = \sqrt{36 \left(\frac{\Delta d_{\rm a}}{d_{\rm a}}\right)^2 + 9 \left(\frac{\Delta C_{\rm c}(d_{\rm a})}{C_{\rm c}(d_{\rm a})}\right)^2 + 4 \left(\frac{\Delta m}{m}\right)^2 + 9 \left(\frac{\Delta C_{\rm c}(d_{\rm m})}{C_{\rm c}(d_{\rm m})}\right)^2} = \sqrt{36(0.043)^2 + 9(0.025)^2 + 4(0.027)^2 + 9(0.025)^2} \sim 28.4\%.$$
(S9)

S2. Technical specifications of the GDI engine

The technical specifications of the GDI engine used in the present study are summarized in Table S1.

Table S1: Engine specifications (bTDC means before Top Dead Center and aTDC means after Top Dead Center)

Specification	Value
Cylinder head	Pentroof type
Compression ratio	12.5:1
Bore	82 mm
Stroke	85 mm
Stroke volume	449 cm3
Fuel direct injection system	Central mounted generic six-hole injector
Injection pressure	150 bars
Spark plug location	Exhaust side
Intake valve timing	Open 334° bTDC
	Close 166° bTDC
Exhaust valve timing	Open 154° aTDC
	Close 330° aTDC

S3. Mass-mobility of polydisperse particles

The mass-mobility relationships of non-stripped and stripped polydisperse particles sampled from two engine operating conditions—1200 rpm speed and 12 bar load as well as 2000 rpm speed and 6 bar load—are shown in Figures S1 and S2, respectively. The mass-mobility exponent of non-stripped particles was in the range of $\sim 2.9 - 3.0$, which implies that these particles were nearly spherical. The mass-mobility exponent of stripped particles was in the range of 2.77 - 2.80, which indicates that the soot particles without any semi-volatile coating had a compact structure.



Figure S1: Representative mass-mobility relationship of non-stripped and stripped particles at engine speed of 1200 rpm and load of 12 bar using a DMA to select particles with certain mobility diameters and subsequently measuring their mass using a CPMA.



Figure S2: Representative mass-mobility relationship of non-stripped and stripped particles at engine speed of 2000 rpm and load of 6 bar using a DMA to select particles with certain mobility diameters and subsequently measuring their mass using a CPMA.

References

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